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**DESIGN CRITERIA FOR ELASTOMERIC BEARINGS**  
**Volume III - Program User's Manual**

**Thiokol/Wasatch Division**  
**A Division of Thiokol Corporation**  
**Brigham City, Utah 84302**

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**March 1976**

**Final Report**

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**Prepared for**

**EUSTIS DIRECTORATE**

**U. S. ARMY AIR MOBILITY RESEARCH AND DEVELOPMENT LABORATORY**

**Fort Eustis, Va. 23604**

## **EUSTIS DIRECTORATE POSITION STATEMENT**

The data contained in this report are the results of an effort designed to improve the state of the art of elastomeric bearing design for helicopter rotor head applications. The products of this effort are a design manual and a computer program based on finite-element techniques. The results of this program are contained in the following four volumes:

Volume I - Final Report  
Volume II - Design Manual  
Volume III - Program User's Manual  
Volume IV - Programmer's Manual

Volume I contains the development and background information used in producing the design manual.

Volume II presents design considerations and procedures, bearing applications, methods of analysis, and techniques for predicting bearing performance.

Volumes III and IV contain the computer code and examples of problems showing sample inputs and outputs.

The products of this effort provide a good foundation for building a comprehensive manual and computer code for the design and analysis of elastomeric bearings for helicopter rotor head applications. It was recognized at the onset of this program that both the manual and the code would be first editions. The results of this effort were expected to define areas requiring further development. Further investigations coupled with feedback from users and/or evaluators are expected to provide material for upgrading the content and format of the manual and codes.

Mr. John Sobczak of the Military Operations Technology Division served as project engineer for this effort.

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than .5, but it may be as close to it as desired.

An extensive input module has been included in the program to make it as user oriented as possible. This includes routines that will automatically generate bearing geometry based on basic input parameters.

The basic program is written in FORTRAN IV with some support routines written in IBM 370 Assembler Language.



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## METHOD

The theoretical basis of the program is contained in Volume I and is not repeated in this manual.

Variable dimensions of the grid have been used throughout so that any size problem can be run if sufficient computer size and time are available.

The solution scheme is a Gaussian reduction. This gives good answers with reasonably short run times.

The program may be run as a standard axisymmetric program giving either one or both of the axisymmetric or torsion solutions. Multiple passes may be made through the program inputting the proper load coefficients to gain an asymmetric loading solution. The accumulation routine will accumulate the displacements and strains for all harmonics to give the final solution.

The program is a finite-element program. To solve for the displacements, stresses and strains throughout a body, it is divided into many small pieces called elements by defining a grid network of nodes over a cross section of the body. For each element, a stiffness matrix is defined according to the basic assumptions for that element. A force vector is created by applying the pressures and loads to the nodes. The element stiffness matrices and force vectors are then assembled into a master set of equations, and the equations are solved using a Gaussian reduction technique. This method of solution has been extensively used at Thiokol for many years and has yielded good answers in reasonably short run times.

There are two basic elements available in this program, and one or the other must be selected in the RUN input section. The first is the linear displacement element. The assumption for it is that the displacements are a linear function through the element. This element is used for the basic axisymmetric runs, the asymmetric loading runs and the torsion runs.

The second element is the isoparametric element. It is a quadratic element where the displacements are assumed to be a quadratic function through the element. This element will in general give better results for the same size or a smaller grid. It has not, however, been as extensively used as the linear element.

In the Type of Material section, there are five element types given. Types 1, 2, 4 and 5 are all linear elements and differ only in the number of degrees of freedom they can handle.



## LOAD DEFINITIONS

For the single-pass, axisymmetric run, loads are input directly as they apply to the entire body of revolution.

For asymmetric loading, a Fourier expansion of the form

$$f(\theta) = A_0 + \sum_{n=1}^N A_n \cos n\theta + \sum_{n=1}^N B_n \sin n\theta$$

must be generated, where N is the number of terms needed to accurately represent the load. All loads are defined in terms of this function of theta. The loads are then input with the constant term  $A_0$  in the first pass and either  $A_n$  or  $B_n$  being entered for the n+1st pass.

The assumption is implicit in the program that the loads are symmetric about the  $0^\circ - 180^\circ$  plane through the body. Because of this symmetry, the radial and axial loads can go in only as an even function and the tangential loads only as an odd function. That is, only the A terms can be used for R or Z loads, and only the B terms can be used for the  $\theta$  loads.

The units of the function f and the coefficients A and B must be in pounds per square inch for pressure, or pounds for nodal point load, over the entire surface affected.

Pressure, shear and traction loads act on the entire surface of revolution on which they are applied. That is, the total load is equal to  $\pi DLP$  where D is the diameter, L the length, and P the pressure or shear.

Nodal point forces apply a load directly to a point and are not distributed over any surface. They are a load on a ring as the loaded point is revolved about the axis.

See Appendix B for the description of a program to generate the Fourier coefficients.

## PROGRAM COORDINATES

The normal orientation of the coordinate system is with the R-axis horizontal and the Z-axis vertical. This is referred to here as a right-handed system; that is, R rotated into Z gives a vector out of the paper.

The indexing system of the program can also be layed out as a coordinate system. By definition, a right-handed system is one in which the I axis rotated into the J axis gives a vector out of the paper.

The two systems, coordinate and indexing, must always agree. That is, they must both be either right-handed or left-handed. If one is right and the other left, the program will give negative area errors on the grid and stop execution.

All angles in the system are measured from the positive R axis counterclockwise in the right-handed system. If the system is transformed, this relative orientation must be maintained.

There are "local origins" referred to in the input section. These are simply temporary origins to aid in the designating of that one node or line. The point will be translated back into the global coordinate system by adding the coordinates of the local origin to the coordinates of the point.

## THE LINE GENERATOR

By means of the Line Generator Record, line and arc segments may be generated internally to connect any\* two points with a set of nodes. The following options are available:

- A. Equal interval - In the straight line all intervals between nodes are equal. In an arc the angular intervals are equal.
- B. Square root of  $r$  - Where  $N$  is the total number of intervals and  $i$  goes from 1 to  $N-1$ , each interval is  $(2i-1)/N^2$  of the total length. For a line, this is an increment of length; for an arc, it is an angular interval.
- C. Geometric progression - The increments of length or angle will be in a geometric progression. Any of  $r$  (the common ratio),  $a$  (the first interval), or  $l$  (the last interval) may be specified. In addition, the second node in the line may be specified and  $r$  will be calculated using the interval from the first node to the second node of the line as  $a$ .

Any of these options may be used to generate any line in the figure. The lines may go in either ascending or descending order of  $I$  or  $J$ .

---

\*Either  $I$  or  $J$  must be constant for all the nodes in a line or arc.

## THE GRID GENERATOR

The grid generator used in this program is one developed at Thiokol by Dr. William Cook.\* This is a linear interpolation method which yields a good grid under most conditions regardless of the curvature of the boundaries.

A two-dimensional space is defined in  $\xi$  and  $\eta$ , and the grid boundary nodes are mapped into this space. A linear interpolation scheme is then used to fill in the remainder of the  $\xi, \eta$  space. Two functions,  $f$  and  $g$ , are then defined that map the  $\xi, \eta$  space into the  $X, Y$  space of the grid. These functions are then used to define the remainder of the grid.

There may be any number of major partitions in the grid and any number of subpartitions within each of these.

A major partition is defined as a closed rectangular section of matrix (the figure may be of any shape, but the section of matrix must be capable of being defined by a minimum and maximum of  $I$  and  $J$ ), all of which will be generated by one and only one grid generation record. The subpartitions are closed rectangular sections of matrix lying within the major partition.

All boundaries of each partition must be completely specified, and there must be no nodes specified that do not lie on the boundaries of a partition.

Each major partition must be specified on a Grid Generator Record, but the program will pick up all subpartitions lying within the specified major partition.

Due to the nature of the Cook grid generator, the following restrictions must be observed when using it:

1. All sections of the matrix that are to be generated by it must be rectangular and the boundaries must be completely defined.
2. If interior nodes are specified, they must form rectangular subsections of the section of matrix being generated. There is no limit to the number of such subsections allowed.

---

\*Cook, William A., BODY ORIENTED (NATURAL) CO-ORDINATES FOR GENERATING THREE-DIMENSIONAL MESHES, International Journal for Numerical Methods in Engineering, Vol 8, 1974, pp. 27-43.

## INPUTTING THE PROGRAM

There are several things about the program that you should be aware of as you set up your input. The sections below define both general things to keep in mind and the specific methods of input required for the various options. For additional information, see Vol II of this report.

### I General

The RUN section must always be used to specify both the geometry type and the element type. In this version of the program, the geometry type is always axisymmetric (option 3). The element type can be either linear displacement (option 4) or isoparametric (option 5).

Each node defines an element, except for isoparametric, where it is every other node. The elements are referenced by the node with the smallest indices of the nodes comprising the element. Those nodes which lie on the boundary of the geometry and do not reference a real element are given a type code of 9, which signifies a missing element. This type code of 9 is also used to code an internal element which has no material in it. Type 9 elements require no material code, and this can be set to 0 or left blank.

Plots can be produced quite simply by specifying the orientation desired, the range of indices of the nodes to be plotted, and the Y paper size. The data will then be scaled to fit the paper height, and the same scale factor will be used for the X direction. The actual scale values used, the minimum values and the maximum values, if calculated, will be printed when each plot is produced.

Several input sections use index increments referred to as I1 and J1. The value I1 is an I index increment, that is, we go from I1 to I2 in increments of I1. The value J1 is a J index increment, and we go from J1 to J2 in increments of J1. For example, if we have a grid where we have material 1 in the elements with an even J index and material 2 in the elements with an odd J index, we can apply these in only 2 records by using a J1 of 2. Another example of its use is in the isoparametric input. This is explained more fully in section VII below.

## **II     The Basic Axisymmetric Problem**

This is the case for which there are no iterations and the basic linear displacement element is being used. The RUN section must show this by inputting options 3 and 4.

## **III    Asymmetric Loading**

The asymmetric loading of an axisymmetric body is accomplished by making multiple passes through the program within one run. Each pass has the harmonic coefficients for one term in the Fourier expansion entered as load conditions. Run option 9, asymmetric loading, must be specified in each pass.

The node number, L11 in the General Data, and the highest harmonic allowed, L12 in the General Data, must both be specified in each pass, and the highest harmonic allowed must be nonzero.

The accumulation of the output is necessary for proper interpretation of the results. Each pass will give results that are the maximum for each coordinate direction. These values have meaning only when multiplied by the proper sine and cosine terms and added to the other terms for that node or element.

The undeformed geometry can be plotted at any time in the run. Accumulation and plotting of the deformed geometry must be in the last pass only. Plots can be made of the displaced geometry at any angle  $\theta$  or axial location  $Z$  by accumulating the data where desired and then inputting a plot section immediately after. The plot routines will always plot the last data accumulated. See the sample input for an example of this.

## **IV     Incremental Loading**

When this option is used, the program automatically cycles through as many times as there are load increments. The cumulative totals of the displacements and strains may be printed out for each increment under the control of the Selective print option. The print suppress option (output option 1) and the Selective print section control the output of the final results, and the Incremental loading print flag with the Selective print option controls all others.

All loads and boundary conditions may be applied in increments. The fraction of the load to be applied in each step may be specified, or on a flag the increments will be calculated in an arithmetic progression.

A set of flags is provided so that specified loads may not be applied incrementally.

Each iteration is handled essentially as a separate run with the program looping through the entire solution scheme. The displaced geometry reflecting all the previous loading is used in each iteration so that stiffness due to bending will be taken into consideration.

#### V Large Deformation

The use of this option takes into consideration the nonlinear terms in the strain displacement relationship and iterates for the final solution. You can also enter strain dependent material properties to take into consideration the nonlinear material behavior.

A Convergence criteria and a series of underrelaxation factors must be entered. The underrelaxation factors must be positive values no greater than 1.0, and at least the last two values must be 1.0. The program will iterate until convergence or the maximum number of iterations is satisfied.

As in the incremental loading option, the print suppress flag controls the final output and the large deformation print flag controls all others.

The energy calculation, reaction forces and accuracy check are not valid for this option and are not allowed.

#### VI Incremental Loading and Large Deformation

These two options may be combined to allow a large deformation iteration inside each incremental loading loop. The controls on each remain the same, and the same set of underrelaxation factors will be used for each increment of load.

## VII Isoparametric Elements

The isoparametric element in this program is a quadratic element with three nodes on each side. The geometry is input and generated just as if a linear displacement element were being used. Note, however, that all element boundaries and material boundaries must fall on nodes with odd indices.

The internal handling of this element requires some special consideration in the input. The element type codes and material numbers must only be on the nodes which designate element corners. To achieve this, I1 and J1 in the TYPE section must be input as 2's.

## VIII Torsion

The torsion solution is part of the zero harmonic in the asymmetric loading case, because it is the only element at present that has a theta degree of freedom. A RUN parameter of 9 must then be specified.

There are two different ways that torsion can be run: (1) with the full asymmetric calculation giving all three degrees of freedom, and using a type 4 element; (2) with torsion only, in which case a type 5 element will be used. The type 5 element is faster than the type 4. Accumulation can be made for both elements, or for any combination of 2, 4 or 5 elements, since they are all asymmetric. Note that the torsion loads act on a moment arm equal to the radial coordinate of the point where they are applied.

## IX Stability

This option is used to determine the stability of a flat, vertically stacked bearing. The only input sections required are the General Input, the Run Input, and the Stability input. The values in the General section will not be used, but due to program flow the section must be included. Run option 20 must be specified.

## X Service Life

To determine the service life of a bearing, this program must be used in conjunction with program S3359SL. When a run is made where data needs to be saved for the service life calculations, output option 20 must be specified. The stresses and



strains will then be output on a file whose name is SERVLIFE. Each set of data put on this file will be identified by two integers which are printed on the output: the first is the julian date and the second is the time of day in centiseconds. These will be used to identify the set of data for program S3359SL. These sets can be put on separate reels of tape, or they can be stacked together on one tape by using the proper Job Control Language.

## TIMING

Timing on this program is a problem because so many factors affect the execution speed.

In general the time required for a run is a function of  $I^2$  and  $J$ .

The asymmetric loading problem requires about two times as long for each pass as a simple axisymmetric problem.

Each iteration on iterative runs must be counted as a full execution.

From our experience, the best way to estimate the time is by experience with the guidelines given above. However, the following equation will give a starting point:

$$\text{Time (in minutes)} = F I^2 J + 1.2$$

where  $F = .0012$  for an asymmetric run and  $.0006$  for an axisymmetric run,  $I$  is the  $I$ -dimension of the grid, and  $J$  is the  $J$ -dimension of the grid. This equation is for the IBM 370/155; the factor  $F$  and the constant  $1.2$  may change for other machines.

### RESTRICTIONS

1. The isotropic Poisson's ratio is restricted to  $0 < \nu < .5$ . At least under some circumstances a zero Poisson's ratio will make the stiffness matrix singular.
2. If a node is on the axis of revolution of an axisymmetrically loaded body, the radial displacement of that node must be fixed at zero.
3. At least one axial component of displacement must be specified to avoid rigid-body axial motion. A sliding boundary condition may also fulfill this requirement if it restrains axial motion.
4. In any row of the grid,  $I$  must run consecutively through each value  $IMIN \leq I \leq IMAX$  for that row. Type 9 elements may be used where needed to provide "voids" in the grid.
5. In the input, a maximum of 9 digits is allowed on each side of the decimal point in a number.
6. There is a maximum of 32,767 materials allowed in the program. However, due to the special internal handling, the number of materials used with isoparametric elements is limited to 50.

## VARIABLE DIMENSIONS

To facilitate running of both large and small problems, the working arrays in the program have been variably dimensioned. Two things should be noted here about the program. First, the program will take as much core storage as it is given; that is, on the step resource usage given at the end of the run, the amount of core used and requested will always be the same. Second, the actual amount of working storage used in each phase will be printed out as part of the output.

The table on the following pages gives the maximum core required for a run. There are many factors which affect the core required for any given run, so the first guess should be taken from the table and then adjusted after a complete run has been made. The excess core available for each part of the program is given with the output.

To find the region required for a run, a look in the table for the I and J size of the grid will give the region in K bytes. If an asymmetric loading case, or an isoparametric case is not being run, the region may be reduced by up to one-third.

For a stability calculation, only 190K will be needed for the run.

MAX STORAGE REQUIRED FOR S2356																
J	I-6	I-3	I-10	I-12	I-14	I-16	I-18	I-20	I-22	I-24	I-26	I-28	I-30	I-32	I-34	I-36
5	196	205	215	226	239	254	270	288	307	327	350	373	399	425	454	484
10	196	205	215	226	239	254	270	288	307	327	350	373	399	425	454	484
15	196	205	215	226	239	254	270	288	307	327	350	373	399	425	454	484
20	196	205	215	226	239	254	270	288	307	327	350	373	399	425	454	484
25	196	205	215	226	239	254	270	288	307	327	350	373	399	425	454	484
30	201	206	215	226	239	254	270	288	307	327	350	373	399	425	454	484
35	204	209	215	226	239	254	270	288	307	327	350	373	399	425	454	484
40	207	209	215	226	239	254	270	288	307	327	350	373	399	425	454	484
45	210	212	218	226	239	254	270	288	307	327	350	373	399	425	454	484
50	213	216	218	226	239	254	270	288	307	327	350	373	399	425	454	484
55	217	219	222	226	239	254	270	288	307	327	350	373	399	425	454	484
60	220	223	225	228	239	254	270	288	307	327	350	373	399	425	454	484
65	223	226	229	233	236	239	254	270	288	307	327	350	373	399	425	484
70	226	229	233	236	240	243	254	270	288	307	327	350	373	399	425	484
75	229	233	236	240	243	247	254	270	288	307	327	350	373	399	425	484
80	233	236	240	243	247	251	258	270	288	307	327	350	373	399	425	484
85	236	239	243	247	251	255	259	263	267	271	276	280	288	307	327	350
90	239	243	247	251	255	259	263	267	271	276	280	288	307	327	350	373
95	242	246	250	255	259	263	267	271	276	280	288	307	327	350	373	399
100	245	250	254	258	263	267	271	276	280	288	307	327	350	373	399	425
105	248	253	258	262	267	271	276	280	288	307	327	350	373	399	425	484
110	252	256	261	266	271	276	280	288	307	327	350	373	399	425	484	515
115	255	260	265	270	275	280	285	290	307	327	350	373	399	425	484	515
120	258	263	268	274	279	284	289	294	307	327	350	373	399	425	484	515
125	261	267	272	277	283	288	293	299	307	327	350	373	399	425	484	515
130	264	270	275	281	287	292	298	303	309	327	350	373	399	425	484	515
135	268	273	278	285	291	296	302	308	314	327	350	373	399	425	484	515
140	271	277	283	289	295	301	307	313	319	327	350	373	399	425	484	515
145	274	280	286	292	299	305	311	317	323	330	350	373	399	425	484	515
150	277	283	289	296	303	309	315	322	328	335	350	373	399	425	484	515
155	280	287	293	300	307	313	320	326	333	340	350	373	399	425	484	515
160	283	290	297	304	311	317	324	331	338	345	357	373	399	425	484	515
165	287	294	301	308	315	322	328	335	342	350	362	373	399	425	484	515
170	290	297	304	311	319	326	333	340	347	355	368	380	399	425	484	515
175	293	300	308	315	322	330	337	345	352	360	373	386	399	425	484	515
180	296	304	311	319	326	334	342	349	357	365	378	392	405	425	484	515
185	299	307	315	323	330	338	346	354	362	370	384	398	411	425	484	515
190	302	310	318	326	334	342	350	358	366	375	389	403	417	431	454	484
195	306	314	322	330	338	347	355	363	371	380	395	409	423	438	454	484
200	309	317	326	334	342	351	359	368	376	385	400	415	429	444	459	484
205	312	321	329	338	346	355	364	372	381	390	405	420	436	451	466	484
210	315	324	333	342	350	359	368	377	385	395	411	426	442	457	473	484
215	318	327	336	345	354	363	372	381	390	400	416	432	448	464	479	484
220	321	331	340	349	358	367	377	386	395	405	421	438	454	470	486	502
225	325	334	343	353	362	372	381	390	400	410	427	443	460	476	493	510
230	328	338	347	357	366	376	385	395	405	415	432	449	466	483	500	517
235	331	341	351	360	370	380	390	400	409	420	437	455	472	489	507	524
240	334	344	354	364	374	384	394	404	414	424	443	460	478	496	513	531
245	337	347	357	367	377	387	397	407	417	427	446	464	482	500	518	536
250	341	351	361	371	381	391	401	411	421	431	450	468	486	504	522	540
255	344	354	364	374	384	394	404	414	424	434	453	471	489	507	525	543
260	347	357	367	377	387	397	407	417	427	437	456	474	492	510	528	546
265	350	360	370	380	390	400	410	420	430	440	459	477	495	513	531	549
270	353	363	373	383	393	403	413	423	433	443	462	480	498	516	534	552
275	357	367	377	387	397	407	417	427	437	447	466	484	502	520	538	556
280	360	371	381	391	401	411	421	431	441	451	470	488	506	524	542	560
285	364	375	385	395	405	415	425	435	445	455	474	492	510	528	546	564
290	367	378	388	398	408	418	428	438	448	458	477	495	513	531	549	567
295	371	382	392	402	412	422	432	442	452	462	481	499	517	535	553	571
300	374	385	395	405	415	425	435	445	455	465	484	502	520	538	556	574
305	378	389	399	409	419	429	439	449	459	469	488	506	524	542	560	578
310	381	392	402	412	422	432	442	452	462	472	491	509	527	545	563	581
315	385	396	406	416	426	436	446	456	466	476	495	513	531	549	567	585
320	388	400	410	420	430	440	450	460	470	480	499	517	535	553	571	589
325	392	403	413	423	433	443	453	463	473	483	502	520	538	556	574	592
330	396	407	417	427	437	447	457	467	477	487	506	524	542	560	578	596
335	399	411	421	431	441	451	461	471	481	491	510	528	546	564	582	600
340	403	414	424	434	444	454	464	474	484	494	513	531	549	567	585	603
345	407	418	428	438	448	458	468	478	488	498	517	535	553	571	589	607
350	410	421	431	441	451	461	471	481	491	501	520	538	556	574	592	610
355	414	425	435	445	455	465	475	485	495	505	524	542	560	578	596	614
360	418	429	439	449	459	469	479	489	499	509	528	546	564	582	600	618
365	422	433	443	453	463	473	483	493	503	513	532	550	568	586	604	622
370	426	437	447	457	467	477	487	497	507	517	536	554	572	590	608	626
375	430	441	451	461	471	481	491	501	511	521	540	558	576	594	612	630
380	434	445	455	465	475	485	495	505	515	525	544	562	580	598	616	634
385	438	449	459	469	479	489	499	509	519	529	548	566	584	602	620	638
390	442	453	463	473	483	493	503	513	523	533	552	570	588	606	624	642
395	446	457	467	477	487	497	507	517	527	537	556	574	592	610	628	646
400	450	461	471	481	491	501	511	521	531	541	560	578	596	614	632	650
405	454	465	475	485	495	505	515	525	535	545	564	582	600	618	636	654
410	458	469	479	489	499	509	519	529	539	549	568	586	604	622	640	658
415	462	473	483	493	503	513	523	533	543	553	572	590	608	626	644	662
420	466	477	487	497	507	517	527	537	547	557	576	594	612	630	648	666
425	470	481	491	501	511	521	531	541								

MAX STORAGE REQUIRED FOR 53359

J	1-6	1-8	1-10	1-12	1-14	1-16	1-18	1-20	1-22	1-24	1-26	1-28	1-30	1-32	1-34	1-36	1-38	1-40	1-42	1-44	1-46	1-48	1-50	1-52	1-54
285	363	375	386	398	410	422	434	445	457	470	491	512	533	554	575	596	616	637	658	679	700	721	742	763	784
290	366	378	390	402	414	426	438	450	462	475	496	518	539	560	581	603	624	645	666	688	709	730	752	773	794
295	369	381	394	406	418	430	442	455	467	480	502	523	545	567	588	610	632	653	675	697	718	740	762	784	805
300	372	385	397	410	422	434	447	459	472	485	507	529	551	573	595	617	639	661	683	705	727	749	771	793	815
305	376	388	401	413	426	439	451	464	476	489	511	533	555	577	600	622	644	666	688	710	732	754	776	798	820
310	379	392	404	417	430	443	456	468	481	495	518	541	563	586	609	631	654	677	699	722	745	767	790	813	835
315	382	395	408	421	434	447	460	473	486	500	523	546	569	592	615	638	661	684	707	730	753	776	799	822	845
320	385	398	412	425	438	451	464	477	491	505	529	552	575	599	622	645	668	691	714	737	760	783	806	829	852
325	388	402	415	429	442	455	469	482	496	510	534	558	581	605	628	651	674	697	720	743	766	789	812	835	858
330	392	405	419	432	446	459	473	487	500	515	539	563	587	611	635	658	682	705	729	752	775	798	821	844	867
335	395	408	422	436	450	464	477	491	505	520	545	569	594	618	642	665	689	712	736	759	783	806	829	853	876
340	398	412	426	440	454	468	482	496	510	525	550	574	599	623	647	670	694	718	741	765	788	812	835	859	882
345	401	415	429	444	458	472	486	500	515	530	555	580	604	628	652	676	699	723	747	770	794	817	841	864	888
350	404	419	433	447	462	476	491	505	519	535	561	586	611	636	660	684	708	732	756	779	803	826	850	873	897
355	407	422	437	451	466	480	495	510	524	540	566	592	618	644	670	695	720	745	770	795	819	843	867	891	915
360	411	425	440	455	470	485	499	514	529	545	571	598	624	650	677	703	729	755	781	807	833	859	885	911	937
365	414	429	444	459	474	489	504	519	534	550	577	603	630	657	683	710	737	763	790	817	843	870	897	924	951
370	417	432	447	463	478	493	508	523	538	553	582	609	636	663	690	717	744	771	798	825	852	879	906	933	960
375	420	436	451	466	482	497	512	528	543	560	587	615	642	670	697	724	752	779	807	834	861	889	916	943	970
380	423	439	454	470	486	501	517	532	548	565	593	621	649	678	706	734	762	790	818	846	874	902	930	958	986
385	427	442	458	474	490	505	521	537	553	570	598	626	654	683	711	739	767	795	823	851	879	907	935	963	991
390	430	446	462	478	494	510	526	542	558	575	604	632	660	689	717	746	774	803	831	859	888	916	945	973	1001
395	433	449	465	481	498	514	530	546	562	580	609	638	667	695	724	753	782	811	840	869	897	926	955	984	1012
400	436	452	469	485	502	518	534	551	567	585	614	643	673	702	731	760	789	818	847	876	905	934	963	992	1021
405	439	456	472	489	506	522	539	555	572	590	620	649	679	708	738	767	797	826	856	885	915	944	974	1003	1032
410	442	459	476	493	510	526	543	560	577	595	625	655	685	715	745	775	804	834	864	894	924	954	984	1014	1043
415	446	463	480	497	514	531	547	564	581	600	630	661	691	721	751	782	812	842	872	902	932	962	992	1022	1051
420	449	466	483	500	518	535	552	569	586	605	636	666	697	727	757	787	817	847	877	907	937	967	997	1027	1056
425	452	469	487	504	521	539	556	574	591	610	641	672	703	734	765	796	827	858	889	919	950	981	1011	1042	1072
430	455	473	490	508	525	543	561	578	596	615	646	678	709	740	772	803	835	866	897	929	960	991	1022	1053	1083
435	458	476	494	512	529	547	565	583	601	620	652	684	715	747	779	810	842	874	905	937	969	1000	1031	1062	1092
440	461	479	497	515	533	551	569	587	605	625	657	689	721	753	785	817	849	881	913	945	977	1009	1041	1073	1104
445	465	483	501	519	537	555	574	592	610	630	663	695	727	760	792	825	857	889	922	954	987	1019	1052	1084	1116
450	468	486	505	523	541	560	578	597	615	635	668	701	733	766	799	832	865	897	930	963	996	1028	1061	1094	1127
455	471	490	508	527	545	564	583	601	620	640	673	706	739	773	806	839	872	905	938	971	1004	1037	1070	1103	1136
460	474	493	512	531	549	568	587	606	624	645	679	712	746	779	813	846	880	913	947	980	1013	1046	1079	1112	1145
465	477	496	515	534	553	572	591	610	629	650	684	717	752	786	819	853	887	921	954	988	1021	1054	1087	1120	1153
470	481	500	519	538	557	576	596	615	634	655	690	724	758	792	826	860	894	928	962	996	1030	1063	1096	1129	1162
475	484	503	523	542	561	581	600	619	639	660	695	729	764	799	833	867	901	935	969	1003	1037	1071	1105	1139	1173
480	487	507	526	546	565	585	604	624	644	665	700	735	770	805	840	875	910	945	980	1014	1048	1082	1116	1150	1184
485	490	510	530	549	569	589	609	629	648	670	705	741	776	811	847	882	917	953	988	1022	1056	1090	1124	1158	1192
490	493	513	533	553	573	593	613	633	653	675	711	746	782	818	853	889	924	960	995	1030	1064	1098	1132	1166	1200
495	496	517	537	557	577	597	618	638	658	680	716	752	788	824	860	896	932	968	1003	1038	1072	1106	1140	1174	1208
500	500	520	540	561	581	602	622	642	663	685	721	757	794	831	867	903	940	976	1012	1048	1084	1120	1156	1192	1228
505	503	523	544	565	585	606	626	647	668	690	727	764	800	837	874	911	948	985	1022	1059	1096	1133	1170	1207	1244
510	506	527	548	568	589	610	631	651	672	695	732	769	806	844	881	918	955	992	1029	1066	1103	1140	1177	1214	1251
515	509	530	551	572	593	614	635	656	677	700	738	775	812	850	887	925	962	1000	1037	1074	1111	1148	1185	1222	1259
520	516	537	558	579	599	620	641	662	683	705	743	781	819	857	895	933	971	1009	1047	1084	1121	1158	1195	1232	1269
525	519	540	562	583	604	625	646	667	688	710	748	786	825	863	901	939	977	1015	1053	1091	1129	1167	1205	1243	1281
530	522	544	565	587	608	629	650	671	692	714	752	790	829	868	907	946	985	1024	1063	1102	1141	1180	1219	1258	1297
535	525	547	569	591	613	635	657	679	701	724	762	801	840	879	918	957	996	1035	1074	1113	1152	1191	1230	1269	1308
540	528	550	573	595	617	639	661	683	705	728	766	805	844	883	922	961	1000	1039	1078	1117	1156	1195	1234	1273	1312
545	531	554	576	599	621	643	665	687	709	732	770	809	848	887	926	965	1004	1043	1082	1121	1160	1199	1238	1277	1316
550	535	557	580	602	624	646	668	690	712	735	773	812	851	890	929	968	1007	1046	1085	1124	1163	1202	1241	1280	1319

## FREFRM INPUT

Input to the program is via subroutine FREFRM. This is a record-oriented free-form input routine. It allows data to be entered without regard to card columns and yet be input as individual unique records.

This routine has the following characteristics:

1. Numbers are not restricted to any particular columns on the card.
2. The numbers may be separated with any of the following:
  - a. A comma is a separator.
  - b. The sign of the number is a separator.
  - c. The single quotation mark (') when defining an alpha string is also a separator.
  - d. The end of a card (column 72) is a separator; that is, if no other separator is encountered before the end of a card, that end will terminate the number. Note that an end of card in an alpha string is an error.
  - e. The L of an L-number is a separator.
  - f. A record terminator (;) will terminate the number as well as the record. A record terminator in an alpha string is an error. A string of dissimilar separators is only one separator. For example, a comma followed by an end of card, followed by an L or quote, is still only one separator.
3. L-Numbers may be used to specify the relative location of the value in the record. An L-Number is an integer preceded by an L which sets the counter to a specific location in the record, the first location being 1. Thus the third location is given by L3. The L-Number must be separated from the following value by a valid separator. The number must always follow the L immediately. The presence of anything other than a digit immediately following the L will

cause an error.

4. A logical record must be terminated by a semicolon (;).
5. Alphanumeric data can be entered anywhere in the string but must be set off with single quotation marks ('). The quotation mark is a separator. Two consecutive quotation marks will cause a quotation mark to be entered in the string. There are 8 characters in each location as defined by the L-Numbers.
6. If two commas appear consecutively, no value is input between them. The value previously put in that location will remain there.
7. If two signs appear consecutively, a zero is assumed between them.
8. If a record begins with an alpha character, it will be assumed to be a flag record or a title, and no other data may be entered in it. These records may begin with 'L', which usually signifies an L-Number, if and only if the second character in the string is alphabetic.
9. Comments may be punched in any card by punching an asterisk (\*) followed by any desired comments. The asterisk and all data following it in that card will be ignored.
10. Only the first 72 columns of each card will be used, so columns 73-80 may be used for sequence numbers if desired.
11. Data entered in a location in a record stays in that location in subsequent records until it is changed by entering a new value in that location.
12. A maximum of 9 digits is allowed on each side of the decimal point in a number.



## FLAG AND TITLE RECORDS

Any record which begins with alphabetic data is considered to be a flag or title record. Flag records are used to flag the beginning of each group of records. If the first four columns of the record contain a flag sequence, the record is considered to be a flag record. If not, it is considered to be a title record and its use is determined from its location in the input stream. The record terminator must not appear in the string. A title record may span several cards, but each flag or title record must end with a record terminator. A flag record cannot exceed a single card.

In the following pages, only four characters are given for each flag since this is all that is checked. Additional characters may be added to make a full statement if desired.

## INPUT FOR PROGRAM

As many of the following groups as needed may be entered. Some of the groups are required; others are optional.

The Title records and General Data must be the first two groups input. All other groups may be input in any order, with the only restriction being that of logical sequence of events. Obviously you can not use an item of data until it is input; therefore, Line Generators must follow the Node records that define their end points, etc. In general, any group may be input more than once; however, this practice should be restricted to avoid overly complex input sets.

Each case run through the program is considered to be a separate case, and all information must be entered for each case. This is also true of the various parts of an asymmetric loading run. Except for the accumulation, each case is considered to be completely separate.

Note that only columns 1 through 72 may be used for data.

### Title

As many cards as desired may be entered here, with the last card containing a record terminator. Each title card must start with an alphabetic character.

### General Data - The flag is GENE

L1	Minimum I value
L2	Minimum J value
L3	Maximum I value
L4	Maximum J value
L5-L9	Not used at present
L10	Base temperature
L11	Mode number if asymmetric loading
L12	Highest mode allowed if asymmetric loading

**Run Parameters - The Flag is RUN**

Run options will be selected from the following table by entering the option numbers as a sequence of integers. The blank options represent options not available in this version of the program.

- 1.
- 2.
3.     Axisymmetric geometry
4.     Linear displacement element
5.     Isoparameteric element
- 6.
- 7.
- 8.
9.     Asymmetric loading of an axisymmetric geometry
10.    Mesh only - check input and test the grid.  
Geometry plots will be produced if requested.
- 11.
- 12.
- 13.
- 14.
- 15.
- 16.
- 17.
- 18.
- 19.
20.    Stability calculation

### Output Options - The Flag is OUTPUT

The options are selected by entering the option numbers from the following table. These options apply to the entire output set; if output is desired for specified sets of nodes or elements only, then use the Selective Print option.

If no output options are specified, the output will be the displacements for each node and the stresses and strains for each element.

1. Print Suppress. The displacements and stresses will not print except as specified by the Selective Print option.
2. Punch the deformed grid.
3. Print the deformed grid.
- 4.
- 5.
- 6.
- 7.
8. Accuracy check. Calculate  $(F-KU)$  for each element and print the 10 elements with the highest error.
9. Reaction forces. Print the reaction forces for the nodes with displacement boundary conditions.
- 10.
- 11.
- 12.
- 13.
14. Print the strain energy.
15. Print the element material properties.
16. Print the loads for each element with nonzero loads and the boundary conditions for each node with nonzero boundary conditions.

17.

18.

19.

20. Output data on tape for service life calculation.

Special Points - The Flag is SPEC

This section allows for a maximum of 100 special points that are not part of the grid. Their purpose is to provide convenient points to be used as local origins for nodes, arcs, etc.

L1 K The special point number. It will be referenced by this number whenever it is used.

L2 X or R coordinate

L3 Y,  $\theta$  or Z coordinate

L4 If nonzero the values in L2 and L3 are considered to be R and  $\theta$ , in degrees. If zero, they are considered to be X and Y.

L5 The number of a previously input special point may be input here, and it will be used as a local origin for point K.

Nodes - The flag is NODE

Each of these records defines one geometry point in the grid.

L1 I

L2 J

L3 R(I,J)

L4 Z(I,J) or  $\theta$  in degrees

L5 Spherical flag - Input a +1 if R and  $\theta$  are input instead of R and Z.

- L6            Local origin flag
- 0 - no local origin
- 1 - special point whose number is in L7 is the local origin
- 2 - the node whose indices are given in L7 and L8 will be used as the local origin
- 3 - the coordinates of the local origin are given in L7 and L8
- L7            K, IL or RL depending on L6
- L8            JL or ZL depending on L6

Line Generators - The flag is LINE

Each record generates a sequence of node points between the existing end points. Either  $I1 = I2$  or  $J1 = J2$  must hold.

L1	I1	]	Beginning node of the line.
L2	J1		

L3	I2	]	Ending node of the line.
L4	J2		

- L5            Rotation code
- +1 if rotating from the positive R to the positive Z axis.
- 1 if rotating from the positive Z to the positive R axis.
- L6            Nodal spacing option code
- '0' equal interval option
- '1' the square root of r option
- '2' geometric progression with 'r' specified

'3' geometric progression with 'a' specified

'4' geometric progression with 'r' specified

'5' geometric progression with the second node in the line defined previously and used to define 'a'

L7 This field will contain a, r or i depending on which option was specified in L6.

L8 GMIN -The smallest interval allowed for a geometric progression.

L9 GMAX - The largest interval allowed for a geometric progression.

L10 If the sequence of nodes is to form an arc, the way in which the arc center is given is specified here.

0 - not arc - generate a straight line

1 - the special point whose number is given in L11 is the arc center

2 - the node whose indices are given in L11 and L12 is the arc center.

3 - the center of the arc is at the coordinates given in L11 and L12

L11 K, IC or RC depending on L10

L12 JC or ZC depending on L10

Bearing Generator - The flag is BEAR

This section contains parameters necessary to generate a spherical or conical bearing model only.

The node points will be generated and element type and material codes will be set as specified for the elements generated. Type of Material records do not need to be input for the elements of the bearing.

See Figure 1 for the relative location of the indices.

- L1            The minimum I value in the bearing. This may be any value. Default of 1. IMNB
- L2            The maximum I value in the bearing IMXB
- L3            The minimum J value in the bearing. Default of 1. JMINB
- L4-6          Not used at present.
- L7            IEQU If this is nonzero the nodes will be equally spaced across the bearing in the I direction instead of having the first and last interval half the others.
- L8            Element type - see Type of Material section for valid types.
- L9            +0 if a spherical bearing.  
              +1 if a conical bearing.
- L10           The inside radius of the bearing for spherical bearings only.
- L11           Radial coordinate for node IMNB, JMINB.
- L12           Radial coordinate for node IMNB, JMXB\*.
- L13           Radial coordinate for node IMXB, JMINB.
- L14           Radial coordinate for node IMXB, JMXB\*.
- L15-18        Contain the axial coordinates of the corner nodes. For a spherical bearing their only function is to place the node in the proper quadrant; they need not be accurate since they will be recalculated to be sure that the node lies on the proper circle. For a conical bearing they will be used as input.
- L15           Axial coordinate for node IMNB, JMINB.

\*JMXB is JMAX for the bearing and is internally calculated.



- L16            Axial coordinate for node IMNB, JMXB\*.
- L17            Axial coordinate for node IMXB, JMNb.
- L18            Axial coordinate for node IMXB, JMXB\*.

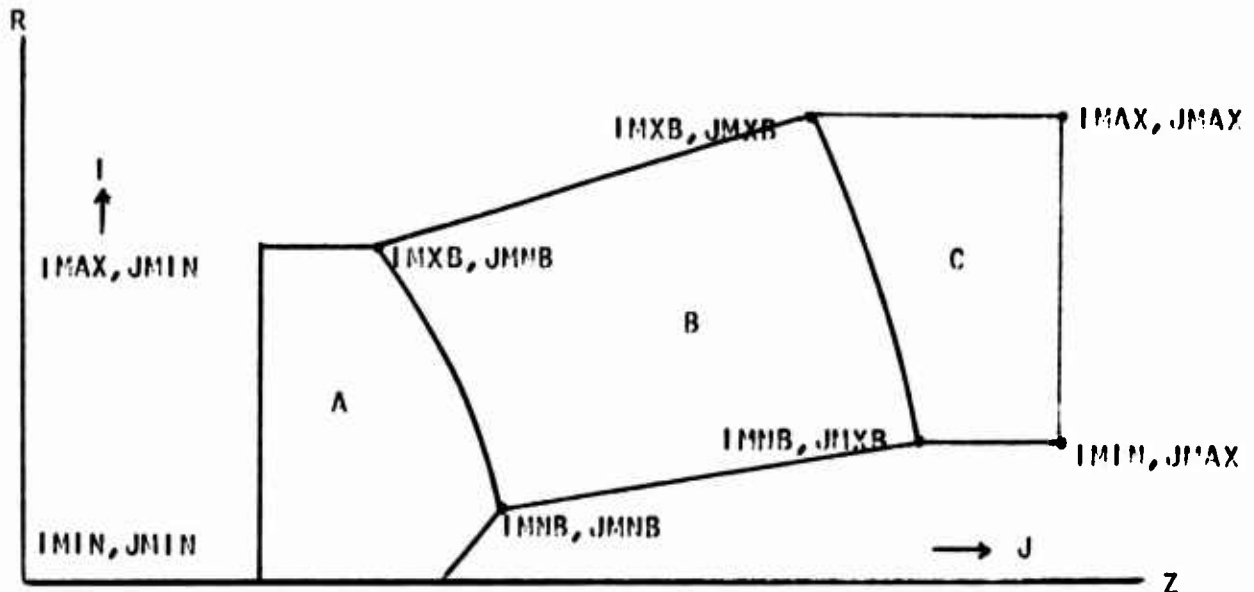


Figure 1. Bearing Example

The complete geometry goes from IMIN,JMIN to IMAX,JMAX. Parts A and C are attachment appliances on the ends of the bearing B. The bearing is defined between indices IMNB, JMNb and IMXB, JMXB\*.

- L19            TS thickness of the shim.
- L20            NS number of columns in one shim.
- L21            ISN number of shims.
- L22            TE thickness of the elastomer.
- L23            NE number of columns in one elastomer.
- L24            IEN number of elastomers.
- L25-29        Not used at present.

\*JMXB is JMAX for the bearing and is internally calculated.

L30 MS material number for shim.  
Default of 1

L31 ME material number for elastomer.  
Default of 2

L32-49 Not used at present.

L50-99 \*\*SHINT shim thickness for each of the shims if not constant. Any value left blank or zero will be replaced by the constant value in L19.

L100-149 \*\*ELAST elastomer thickness for each elastomer if not constant. Any value left blank or zero will be replaced by the constant value in L22.

L150-199 \*\*MATS material number for the shims if not constant. Any value left blank or zero will be replaced by the constant value in L30.

L200-249 \*\*MATE material number for the elastomers if not constant. Any value left blank or zero will be replaced by the constant value in L31.

Grid Generator - The flag is GRID

L1	I1	}
L2	J1	
L3	I2	
L4	J2	

These specify the region to be generated. The generator used is the Cook grid generator, and the region specified must be a closed rectangular system in the I, J plane.

\*\*Starting with JMN8 and proceeding in order out to JMXB

Type of Material - The flag is TYPE

Each of these records assigns a material number and an element type to a block of elements. A material property table must be input for each number assigned here. Note that these are element indices.

L1	I1
L2	J1
L3	I2
L4	J2
L5	Element type

The valid element types are:

- 1 Axisymmetric with R and Z degrees of freedom
- 2 Axisymmetric with R,  $\theta$  and Z degrees of freedom. Used for asymmetric loading.
- 3 Isoparametric quadratic element with three nodes per side and R and Z degrees of freedom.
- 4 Element to be used on mode 0 of an asymmetric loading run when theta loads are used. Torsion in connection with radial and axial loads.
- 5 Element to be used on mode 0 of an asymmetric loading run when torsion is the only load. This element is much faster than type 4.
- 9 Null or void element. This type must be assigned to the last node in each row and column. It may also be used to define holes in the material. It is automatically assigned to the nodes on IMAX and JMAX.

L6	Material number
L7	I1 - if input, I will go from I1 to I2 in increments of I1
L8	J1 - if input, J will go from J1 to J2 in increments of J1

**Boundary Conditions - The flag is BOUN**

Each record assigns boundary conditions to a set of nodes.

L1	I1
L2	J1
L3	I2
L4	J2
L5	R boundary condition code*
L6	R boundary condition value or harmonic coefficient
L7	$\theta$ boundary condition code*
L8	$\theta$ boundary condition value or harmonic coefficient
L9	Z boundary condition code*
L10	Z boundary condition value or harmonic coefficient
L11-12	not used at present
L13	Sliding boundary condition code, 0 or 1 only.
L14	Angle in degrees of the line the node must slide on. See Figure 2.
L15	I1 if input, I will go from I1 to I2 in increments of I1
L16	J1 if input, J will go from J1 to J2 in increments of J1

**\*The valid codes are:**

- 0 - no boundary condition
- 1 - displacement boundary condition
- 2 - A nodal force is to be applied. Nodal forces are the forces on the entire circumference.

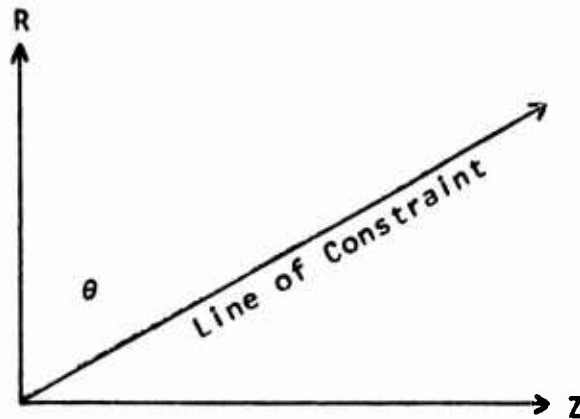


Figure 2. Sliding Boundary Condition

**Pressures** - The flag is PRES

Each record puts pressure along a line of nodes.  $P$  is normal to the element surface and  $S$  is parallel to it. The values  $P_R$ ,  $P_{TH}$  and  $P_Z$  are traction loads in the coordinate system and will decompose into pressure and shear components internally.

L1            I1

L2            J1

L3            I2

L4            J2

L5            Direction code:

0 if the pressure is to be applied to the elements to the left of the line moving from point I1, J1 to point I2, J2.

1 if the pressure is to be applied to the elements on the right.

See Figures 3 and 4.

**Note:** This code is correct for a right-handed indexing system. For a left-handed indexing system, the codes must be interchanged. That is, 0 is pressure to the right and 1 to the left.

L6	P1	Pressure at node I1, J1
L7	P2	Pressure at node I2, J2
L8	S1	Shear at node I1, J1
L9	S2	Shear at node I2, J2
L10	PR1	Radial traction at node I1, J1
L11	PR2	Radial traction at node I2, J2
L12	PTH1	Theta traction at node I1, J1
L13	PTH2	Theta traction at node I2, J2
L14	PZ1	Axial traction at node I1, J1
L15	PZ2	Axial traction at node I2, J2

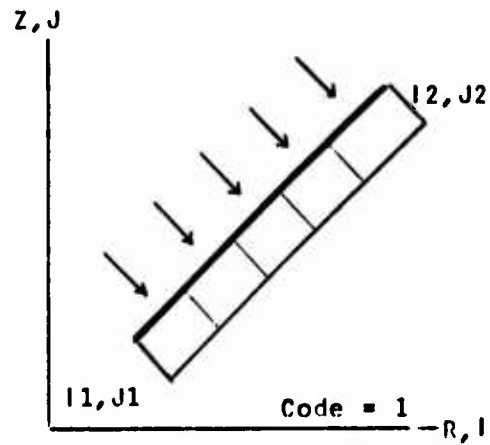
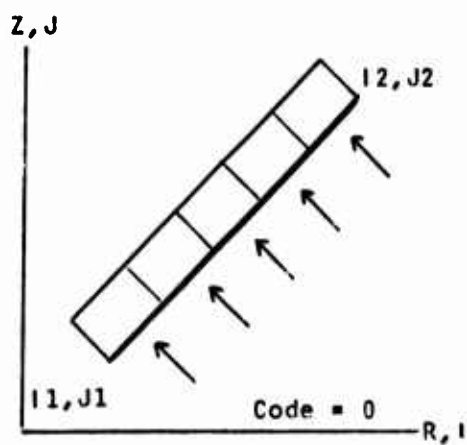


Figure 3. Right-handed Pressure Convention

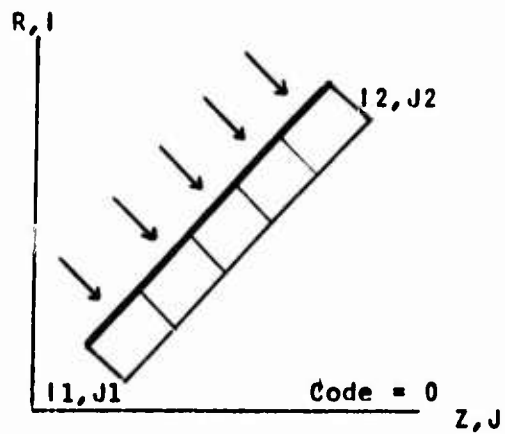
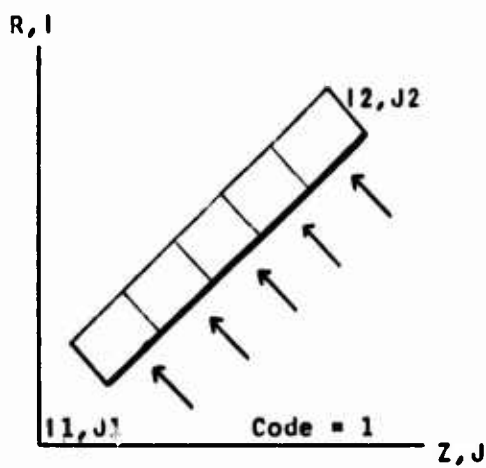


Figure 4. Left-handed Pressure Convention

Body Force Loads - The Flag is BODY

L1	I1	
L2	J1	
L3	I2	
L4	J2	
L5	Radial	body force harmonic coefficient ( $\rho\omega^2$ )
L6	Theta	body force harmonic coefficient (F)
L7	Axial	body force harmonic coefficient (F/L <sup>3</sup> )

Temperature Distribution - The Flag is TEMP

There are two ways in which a temperature distribution may be applied in this program. Only one form may be used in any one run.

A. Constant temperature over the whole body.

The first record must be the word CONSTANT enclosed in single quotation marks. This record must be followed by one record of the following form:

L1 The constant temperature

B. Radius vs temperature tables

The first record must be the character string R vs T enclosed in single quotation marks. This record will be followed by as many records of the following form as necessary to completely define the temperature of each element:

L1	I1	] These are node point	
L2	J1		indices which define
L3	I2		the area over which



L4	J2	└ this table applies.
L5	R1	
L6	T1	Temperature at radius R1
L7	R2	
L8	T2	Temperature at radius R2

**Material Properties** - The flag is MATE

This program accepts isotropic materials only.

For nonstrain or temperature-dependent materials, only one record is entered for each material.

For materials that are strain or temperature dependent, several records will be entered for each material. One record must be entered for each entry in the table. The material number and name need not be entered on each record of the table. If the table is strain dependent, the strain value in the first record must be nonzero and the table must be so input that a zero strain value can be found. If the table is temperature dependent, the temperature value in the first record must be nonzero.

L1	Material number
L2	Young's modulus (E) or large deformation K1*
L3	Poisson's ratio ( $\nu$ ) or large deformation K2*
L4	Alpha for thermal loading
L5	Temperature for which these material properties apply if they are temperature dependent.
L6	Strain at which these material properties apply if they are strain dependent (large deformation).
L7-9	A 24-character material name enclosed in quotation marks.

---

\*See Vol II for definition.

NOTE: Due to special internal handling of material properties, do not use L-Numbers in the input. Always specify missing parameters by commas.

Selective Print Option - The Flag is SELE

This section may be used to ask for output of displacements, stresses and material properties if the overall output of those items has been suppressed in the Output section. As many records as desired may be entered.

L1	I1
L2	J1
L3	I2
L4	J2

Incremental Loading - The Flag is INCR

This section will cause the program to iterate, applying a portion of the load each time and accumulating the results.

L1	The number of increments (N) over which the loads will be applied. If this is positive, then the second record for this set must be entered with N fractions on it. If negative, the loads will be applied in an arithmetic progression.
L2	If a nonzero value is input, printout of material properties, displacements and stress/strains will be made each iteration for those nodes specified by the selective print. The complete output will be printed on the final increment unless the print suppress flag, output option 1, is turned on.

The following set of flags tells which loads will be applied incrementally. If the flag is zero, the load will be applied in increments; if nonzero the full load, if any, will be applied in each iteration.

L3	Pressure and traction loads PR, PTH and PZ
----	--

L4	Shear
L5	Temperature
L6	R body force
L7	Z body force
L8	R nodal load
L9	Z nodal load
L10	Nodal displacements

Incremental loading fraction record

This record must be entered if N is positive and may be entered to override calculated values if N is negative.

L1	Fraction for 1st Increment
L2	Fraction for 2nd Increment
	etc.

for N (max. of 100) increments

Large Deformation - The Flag is LARG

This section controls the large deformation iteration. One record of the following form must be entered.

L1	The number of the last iteration to be allowed.
L2	If nonzero, printout of displacements will be made each iteration for those nodes specified by the Selective print option.
L3	Convergence factor. Convergence is attained when the absolute value of the largest change in displacements is less than this value.
L4-103	The underrelaxation factors for iterations 1-99. The default value is 1.0.

### Geometry Plots - The Flag is GEOM

This section is used to request the plotting of the geometry, either deformed or undeformed. Plots for which the displacement multiplier is zero will be produced during the input phase at the time the plot section is encountered. It is possible to plot the geometry at any point in its development by putting geometry plot sections in at the desired points, such as after the node records, etc.

Note that data will not carry over from one record to the next in this section.

- |    |  |  |
|----|--|--|
| L1 | Axis parameter option  |  |
|    | 1 if the R-axis is to be horizontal to the right   |  |
|    | 2 if the Z-axis is to be horizontal to the right   |  |
| L2 | I1   | ] These indices define the region of the geometry to be plotted. |
| L3 | J1   |  |
| L4 | I2   |  |
| L5 | J2   |  |
| L6 | The displacement multiplier. If zero, the original geometry will be plotted. If nonzero, the displacements will be multiplied by this before being added to the node coordinates for plotting. |  |
| L7 | X-axis scale factor*   |  |

\*The X and Y axes here refer to the paper rather than the geometry. Not all of these parameters need be entered. The maximum and minimum values will be obtained from the data if not specified. Also, both the scale factors and the paper size need not be specified. The relation  $\text{Scale} = \text{paper size} / (\text{Max} - \text{Min})$  will be used to find any missing data. The scale factor will always be used if entered. If the scale factors are not input, they will be calculated to keep the plot on the paper. If the X paper size and scale factor are not entered, the Y scale factor will be used for both axes.

L8	Y-axis scale factor*
L9	Minimum value on the X-axis*
L10	Maximum value on the X-axis*
L11	Minimum value on the Y-axis*
L12	Maximum value on the Y-axis*
L13	X-axis paper size*
L14	Y-axis paper size*
L15	The X location in inches of the beginning of the title.
L16	The Y location in inches of the beginning of the title.
L17	The letter size for the title in inches.
L18-19	Not used at present
L20-25	Up to 48 characters of plot title information as an alpha string enclosed in quotation marks.

Accumulation Control - The Flag is ACCU

This section must be entered only on the last case of an asymmetric loading run. Under control of these records the displacements and strains will be read in for each mode and accumulated. The accumulated output will be printed. It may also be plotted by putting a geometry plot request after the desired displacements have been calculated; thus there may be several Accumulation and Plot sections alternating in the input.

---

\*The X and Y axes here refer to the paper rather than the geometry. Not all of these parameters need be entered. The maximum and minimum values will be obtained from the data if not specified. Also, both the scale factors and the paper size need not be specified. The relation  $\text{Scale} = \text{paper size} / (\text{Max} - \text{Min})$  will be used to find any missing data. The scale factor will always be used if entered. If the scale factors are not input, they will be calculated to keep the plot on the paper. If the X paper size and scale factor are not entered, the Y scale factor will be used for both axes.

Note that to plot a transverse cut, I goes from 1 to the maximum number of nodes in the Radial direction and J goes from 1 to 19.

L1            Option Flag

1    Output will be an axial cut at angle  $\theta$ .

2    Output will be a transverse cut at the axial distance Z.

L2             $\theta$  or Z depending on L1.

Stability - The flag is STAB

This section is used to input the data for the stability calculation. Run option 20 must be specified for the calculations to be made. The stack is assumed to be a vertical stack beginning and ending with a rubber pad; hence the number of shims is assumed to be one less than the number of rubber pads.

L1            The thickness of each rubber pad.

L2            The thickness of each shim.

L3            The inside diameter of the stack if it is cylindrical.

L4            The outside diameter.

L5            The number of rubber pads.

L6            The Young's modulus for the rubber pads.

L7            The convergence accuracy factor. When  $(1 - \text{old buckling load} / \text{new buckling load})$  is less than this value, the iteration has converged.

L8            The maximum number of iterations to be allowed.

End of Harmonic - The flag is END

An end record must terminate the input for each portion of the run. This signals the program to stop reading and solve the problem as defined.

## SAMPLE INPUT

Two cases are shown here. They are illustrations only and do not necessarily reflect real problems.

Sample case 1 is a flat bearing. It is set up in a right-handed system. Figure 5 is a plot of the geometry, and arrows have been drawn in to show the R, Z, I and J axes. This is purely a bearing with no additional parts added. The output for this case is not given in this document since its form is similar to that of case 2.

Sample case 2 is a spherical bearing with a steel attachment on one end. It is oriented in a left-handed system. Figure 6 is a plot of the geometry, and arrows have been drawn in to show the axes. Note how the input sections are ordered to achieve the desired results. The output from this case is given here to show its form.

# SAMPLE PROBLEM 1

## SAMPLE PROBLEM 1 ;

### GENERAL DATA ;

1, 1, 6, 14 ;

### RUN ;

3, 4 ;

### OUTPUT ;

8, 9, 14 ;

### BEARING ;

1, 6, 1, 18, 1, 1, C, C, 3.C, 3.0, 0, 2.0, 0, 2.C,  
.55, 2, 2, .3, 3, 3 ;

### BOUNDARY COND ;

1, 1, 1, 14, 1, 0 ;

1, 1, 6, 1, 0, 0, 0, C, 1, 0 ;

### BOCY FORCE ;

1, 1, 6, 4, 0, C, 10. ;

1, 6, 6, 9 ;

1, 11, 6, 14 ;

1, 4, 6, 6, 0, 0, 30. ;

1, 9, 6, 11 ;

### MATERIALS ;

1, 1.OE6, .3, , , ' STEEL SHIM' ;

2, 300., .49999, , , 'RUBBER PAC' ;

### GEGMETRY PLCT ;

1, 1, 1, 6, 14, 0, 3., 3., -.333, , -.333, , , 1.C, 8.5, .12,

L20 'SAMPLE 1' ;

1, 1, 1, 6, 14, 5.0, 4., 4., C, C, , , 1.0, 8.5, .12,

L20, 'DEFORMED SAMPLE 1' ;

ENC ;



[illegible]

◆

▲

**Figure 5. Sample Problem 1**

## SAMPLE PROBLEM 2

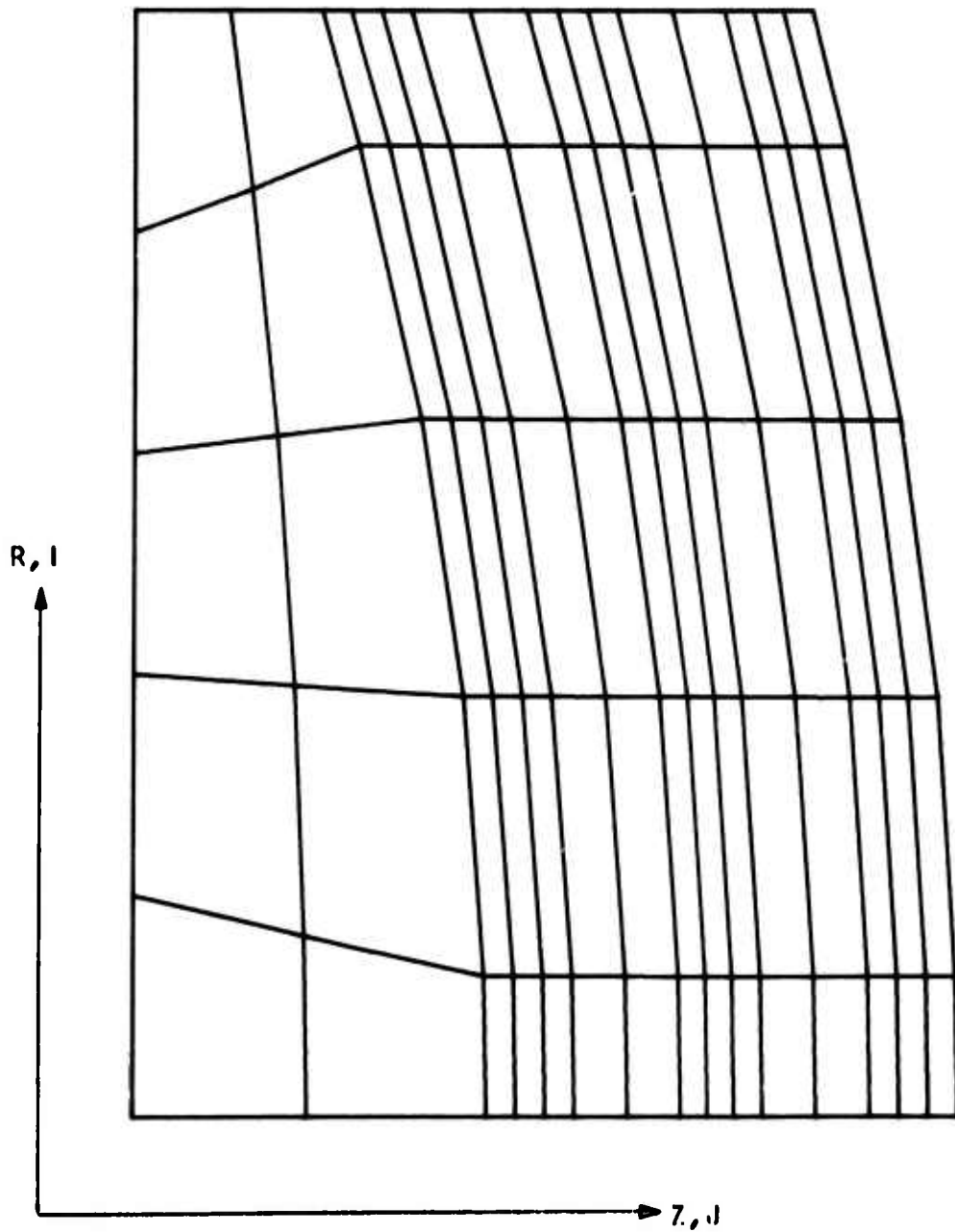
```

      SAMPLE PROBLEM 2,   ZEROETH HARMONIC ;
GENERAL DATA ;
  1, 1, 6, 16, 111, 0, 1 ;
RUN ;
  3, 4, 9;
NODES ;
  1, 1, 0.0, 9.0 ;
  6, 1, 3.0, 9.0 ;
BEARING ;
  1, 6, 3, 18, 2, 0, 10.C, C, C, 2.0, 3.C, 10.C, 12.C, 9.0, 11.0,
    .3, 2, 2, .24, 3, 3;
LINES ;
  1, 1, 6, 1 ;
  1, 1, 1, 3 ;
  6, 1, 6, 3 ;
GRID ;
  1, 1, 6, 3 ;
TYPE ;
  1, 1, 5, 2, 2, 1 ;
BOUNDARIES ;
  1, 2, 1, 16, 1, 0 ;
  1, 1, 6, 1, 1, 0, 1.C, 1, 0 ;
PRESSURE ;
  1, 16, 6, 16, 1, 50., 50. ;
MATERIALS ;
  1, 1.0E6, .3,,,, 'STEEL' ;
  2, 300., .49999,,,, 'RUBBER' ;
GEOMETRY PLCT ;
  2, 1, 1, 6, 16, C, 2.0, 2.C, 8.C, -1.C,,, 2.C, 3.3, .12 ,
    L2C, 'SAMPLE 2 UNDEFORMED';
END OF HARMONIC ;
      SAMPLE PROBLEM 2,   FIRST HARMONIC ;
GENERAL DATA ;
  1, 1, 6, 16, 111, 1, 1 ;
RUN ;
  3, 4, 9;
NODES ;
  1, 1, 0.0, 9.C ;
  6, 1, 3.0, 9.C ;
BEARING ;
  1, 6, 3, 18, 2, 0, 10.C, C, C, 3.C, 3.C, 10.C, 12.C, 9.0, 11.0,
    .3, 2, 2, .24, 3, 3;
LINES ;
  1, 1, 6, 1 ;
  1, 1, 1, 3 ;
  6, 1, 6, 3 ;

```

GRID ;  
 1, 1, 6, 3 ;  
 TYPE ;  
 1, 1, 5, 2, 2, 1 ;  
 BOUNDARIES ;  
 1, 1, 6, 1, 1, 0, 1, 0, 1, 0 ;  
 PRESSURE ;  
 6, 1, 6, 16, 0, 100., 100. ;  
 MATERIALS ;  
 1, 1.0E6, .3, , , , 'STEEL' ;  
 2, 300., .49999, , , , 'RUBBER' ;  
 ACCUMULATION ;  
 1, 0 ;  
 GEOMETRY PLOT ;  
 2, 1, 1, 6, 16, 5, 2.0, 2.0, 8.0, -1.0, , , 2.0, 8.3, .12 ,  
 L20, 'SAMPLE 2 DEFORMED 5X' ;  
 END OF HARMONIC ;

**SAMPLE 2 UNDEFORMED**



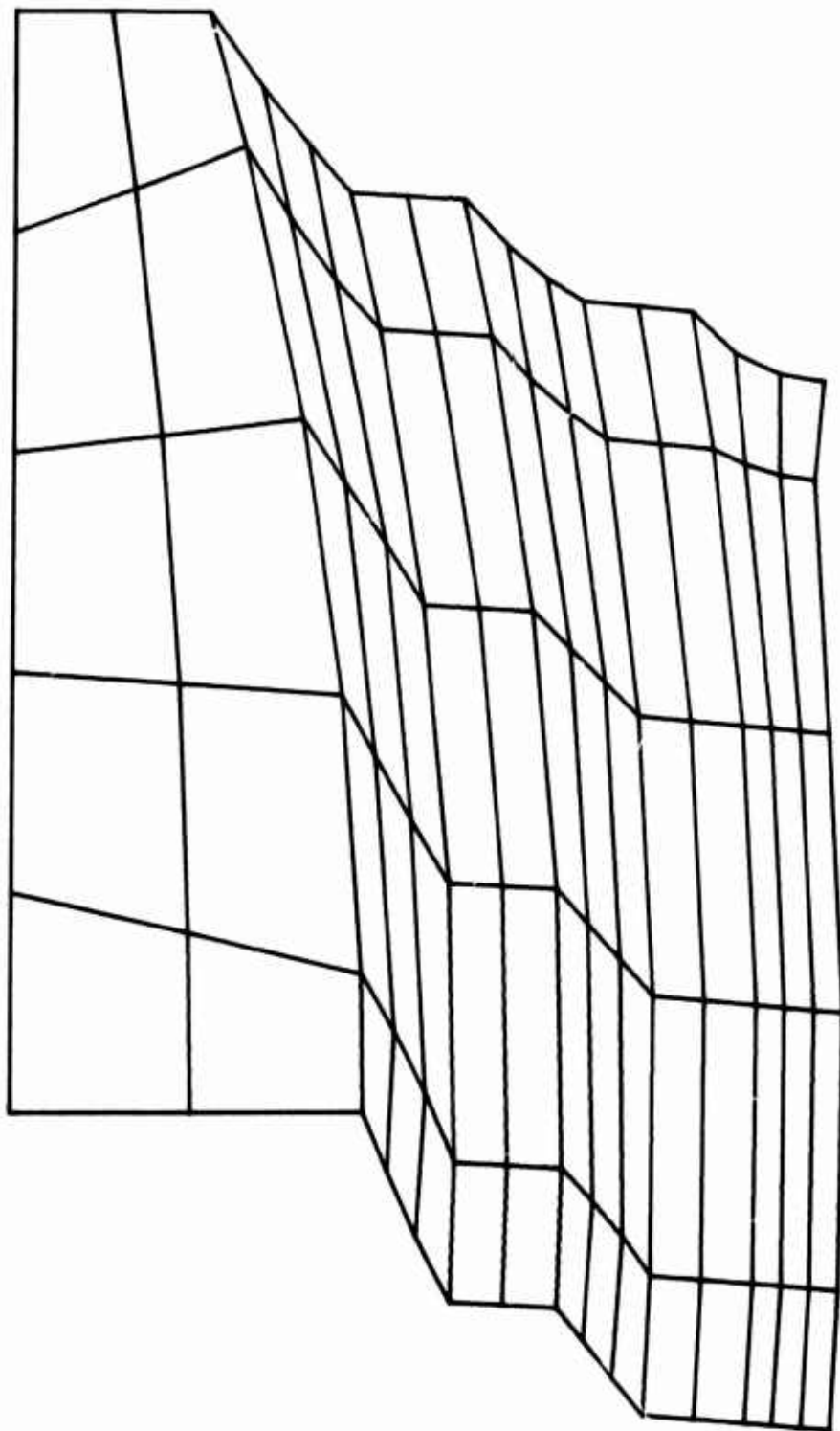
**Figure 6. Sample Problem 2**

## SAMPLE OUTPUT

The output shown is for Sample Input Case 2. Some of the pages are combined to save space, and in some cases the full output is not shown. Note that the input for the second pass has been heavily abbreviated since most of it is the same as for the first pass.

Figure 7 is a plot of the deformed geometry with the displacements multiplied by 5.

**SAMPLE 2 DEFORMED 5X**



**Figure 7. Deformed Sample 2**

# THICKOL AUTOMATED STRESS SYSTEM

## SAMPLE PROBLEM 2, ZEROTH HARMONIC

DATE 8/20/75 TIME 08:49 PAGE 1

PROG S3359 SAMPLE PROBLEM 2, ZEROTH HARMONIC

\*\*\* GENERAL DATA \*\*\*

THE GENERAL DATA VALUES ARE:  
 1.000000000000  
 0.0  
 0.0  
 1.000000000000  
 0.0  
 0.0  
 16.000000000000  
 0.0  
 1.000000000000

THE GENERAL DATA VALUES ARE:

INIT 1  
 JMIN 1  
 JMAX 6  
 JMAX 16  
 BASE TEMPERATURE = 0.0  
 MODE = 0 MAX MODE = 1

\*\*\* RUN PARAMETERS DATA \*\*\*

THE RUN FLAGS ARE:  
 3.0 4.0 5.0

THE RUN PARAMETER OPTIONS THAT ARE IN EFFECT FOR THIS CASE ARE:

ASYMMETRIC  
 LINEAR DISPLACEMENT ELEMENT  
 ASYMMETRIC LOADING

DATE 8/20/75 TIME 08:49 PAGE 2

PROG S3359 SAMPLE PROBLEM 2, ZEROTH HARMONIC

I J NODE INPUT RECORDS Z OR THETA POL IL OR K JL ZL

1 1 0.0 9.00000000 CO 0  
 6 1 3.00000000 CO C

BEARING GENERAL PARAMETERS

JMIN = 1 JMAX = 6  
IVEC = 0 VANG = 0.0

ELEMENT TYPE FOR BEARING ELEMENTS IS 2

A SPHERICAL BEARING WILL BE GENERATED.

GEOMETRY

RI = 1.000000 OI RF = 1.132000 OI  
RIF = 0.0  
ZII = 0.000000 OI ZIF = 1.200000 OI  
TS = 3.000000-01 MS = 2  
TE = 2.000000-01 ME = 3

MATERIAL PROPERTIES

MS = 1 ME = 2

LINE GENERATOR RECORDS

11	J1	12	J2	RCO	OPT	XIN	GWIN	GMAX	IC	JC	RC	ZC
1	1	6	1	0.	0	0.0	0.0	0.0				
1	1	1	3	0.	0	0.0	0.0	0.0				
6	1	6	3	0.	0	0.0	0.0	0.0				

GRID GENERATOR RECORDS

J1 J2

PARTITION GENERATED FROM 1 1 TO 6 3

MATERIAL TYPE RECORDS

11	J1	12	J2	TYPE	MAT	I-INC	J-INC
1	1	5	2	2	1	1	1





GRID COORDINATES AND PARAMETERS

I	J	R	Z	TYPE	MAT	FLG	RADIAL	AXIAL	BOUNDARY CONDITIONS		SLIDING	
									ROTATION	THETA		
1	1	0.000000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
1	2	0.000000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
3	1	1.200000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
4	1	1.800000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
5	1	2.400000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
6	1	3.000000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
1	2	0.000000	9.000000	2	1	0	1	0.0	1	0.0	0	0.0
2	2	0.491090	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
3	2	1.170125	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
4	2	1.840351	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
5	2	2.517511	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
6	2	3.000000	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
1	3	0.000000	9.000000	2	2	0	1	0.0	0	0.0	0	0.0
2	3	0.380774	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
3	3	1.140113	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
4	3	1.892840	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
5	3	2.634589	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
6	3	3.000000	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
1	4	0.000000	9.000000	2	2	0	1	0.0	0	0.0	0	0.0
2	4	0.380774	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
3	4	1.130664	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
4	4	1.892840	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
5	4	2.634589	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
6	4	3.000000	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
1	5	0.000000	9.000000	2	2	0	1	0.0	0	0.0	0	0.0
2	5	0.380774	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
3	5	1.130664	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
4	5	1.892840	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
5	5	2.634589	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
6	5	3.000000	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
1	6	0.000000	9.000000	2	1	0	1	0.0	0	0.0	0	0.0
2	6	0.380774	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
3	6	1.130664	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
4	6	1.891981	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
5	6	2.634128	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
6	6	3.000000	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
1	7	0.000000	9.000000	2	1	0	1	0.0	0	0.0	0	0.0
2	7	0.380331	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
3	7	1.130956	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
4	7	1.891475	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
5	7	2.633857	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
6	7	3.000000	9.000000	2	1	0	0	0.0	0	0.0	0	0.0
1	8	0.000000	9.000000	2	2	0	1	0.0	0	0.0	0	0.0
2	8	0.380175	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
3	8	1.130546	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
4	8	1.890553	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
5	8	2.633598	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
6	8	3.000000	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
1	9	0.000000	9.000000	2	2	0	1	0.0	0	0.0	0	0.0
2	9	0.380094	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
3	9	1.130335	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
4	9	1.890744	9.000000	2	2	0	0	0.0	0	0.0	0	0.0
5	9	2.633464	9.000000	2	2	0	0	0.0	0	0.0	0	0.0

SAMPLE PROBLEM 2 - ZEROTH HARMONIC			GRID COORDINATES AND PARAMETERS										DATE 8/20/75 TIME 06:49 PAGE 12			
I	J	R	Z	TYPE	PAT	FLG	PRINT	RADIAL	AXIAL	BOUNDARY CONDITIONS		SLIDING				
										ROTATION	THETA					
6	9	3.000000	10.127463	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	10	0.000000	10.700000	2	2	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	10	C.380016	10.653250	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	10	1.138129	10.632298	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	10	1.890501	10.531667	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	10	2.633334	10.370895	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	10	3.000000	10.270832	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	11	C.090000	10.780000	2	1	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	11	C.379539	10.773302	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	11	1.137928	10.719772	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	11	1.890263	10.612578	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	11	2.633207	10.453450	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	11	3.000000	10.354149	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	12	C.090000	10.930000	2	1	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	12	C.375799	10.623399	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	12	1.137563	10.870642	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	12	1.890833	10.763361	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	12	2.632975	10.608126	2	1	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	12	3.000000	10.510228	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	13	C.090000	11.080000	2	2	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	13	C.374666	11.073493	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	13	1.137214	11.021486	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	13	1.894421	10.917714	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	13	2.632754	10.762667	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	13	3.000000	10.666133	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	14	C.090000	11.160000	2	2	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	14	C.379597	11.153542	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	14	1.137033	11.101926	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	14	1.895208	10.955931	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	14	2.632640	10.845036	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	14	3.000000	10.745214	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	15	C.090000	11.240000	2	2	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	15	C.379529	11.233591	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	15	1.136897	11.182359	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	15	1.890000	11.080130	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	15	2.632528	10.921369	2	2	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	15	3.000000	10.822448	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
1	16	C.090000	11.320000	9	C	0	0	1.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
2	16	C.375463	11.313638	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
3	16	1.136685	11.262786	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
4	16	1.891757	11.161310	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
5	16	2.632419	11.009667	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0
6	16	3.000000	10.918237	9	C	0	0	0.0	0.0	0.0	0.0	0.0	0	0	0.0	0.0

SAMPLE PROBLEM 2. ZEROTH HARMONIC

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DISPLACEMENTS FOR ASYMMETRIC LOADING				MODE -	THETA-DISPLACEMENT	Z-DISPLACEMENT
R-COORDINATE	Z-COORDINATE	R-DISPLACEMENT				
0.00000	9.00000	0.0		0.0		0.0
0.00000	9.00000	0.0		0.0		0.0
1.20000	9.00000	0.0		0.0		0.0
1.00000	9.00000	0.0		0.0		0.0
2.40000	9.00000	0.0		0.0		0.0
3.00000	9.00000	0.0		0.0		0.0
0.00000	9.50000	0.0		0.0		0.0
0.49109	9.49203	2.794030-07		0.0		-2.889120-05
1.17613	9.46451	2.194370-06		0.0		-2.851180-05
1.84635	9.41250	5.276470-06		0.0		-2.554860-05
2.51751	9.33625	8.114020-06		0.0		-1.846650-05
3.00000	9.26970	9.734930-06		0.0		-8.920870-06
0.00000	10.00000	0.0		0.0		-6.584140-06
0.38077	9.99275	-1.818240-06		0.0		-2.618540-05
1.14011	9.93479	-3.208460-06		0.0		-5.619190-05
1.84284	9.81922	2.302060-07		0.0		-5.261260-05
2.63459	9.64671	9.657650-06		0.0		-3.703230-05
3.00000	9.53535	1.290790-05		0.0		-1.720160-05
0.00000	10.00000	0.0		0.0		-1.173250-05
0.38068	10.07281	-1.769220-04		0.0		1.929180-04
1.13566	10.01534	2.242250-04		0.0		-1.227510-04
1.89235	9.50074	1.669580-03		0.0		-5.244600-05
2.63443	9.72565	4.059290-03		0.0		-5.141340-04
3.00000	9.62322	4.972830-03		0.0		-1.226410-03
0.00000	10.16000	0.0		0.0		-1.774360-03
0.38059	10.15287	-1.716120-04		0.0		4.348690-05
1.13462	10.09582	1.873280-04		0.0		6.003210-05
1.85226	9.98223	1.654330-03		0.0		-2.544570-04
2.63428	9.81255	3.790250-03		0.0		-7.196600-04
3.00000	9.70899	5.508480-03		0.0		-1.646360-03
0.00000	10.24000	0.0		0.0		-2.305520-03
0.38050	10.23293	-2.797010-05		0.0		1.583090-04
1.13438	10.17641	-4.137730-06		0.0		6.115150-05
1.85158	10.06370	1.634860-05		0.0		-2.845360-04
2.63413	9.89540	1.467510-05		0.0		-6.190070-04
3.00000	9.79065	2.690640-06		0.0		-6.620690-04
0.00000	10.39000	0.0		0.0		-5.503560-04
0.38033	10.38304	2.742310-05		0.0		1.543440-04
1.13496	10.32736	5.975730-05		0.0		4.956180-05
1.85148	10.21636	6.693390-05		0.0		-2.584760-04
2.63386	10.05062	5.201820-05		0.0		-6.310650-04
3.00000	9.94747	3.411560-05		0.0		-6.686510-04
0.00000	10.54000	0.0		0.0		-5.533260-04
0.38017	10.53314	8.160040-05		0.0		1.242770-04
1.13455	10.47833	1.233350-04		0.0		2.979040-05
1.85059	10.36698	1.184940-04		0.0		-3.145830-04
2.63340	10.20567	9.225590-05		0.0		-6.438430-04
3.00000	10.10404	6.996330-05		0.0		-6.762970-04
						-9.573370-04

DISPLACEMENTS FOR ASYMMETRIC LOADING				DATE 8/20/75 TIME 08:49 PAGE 14	
I	J	R-COORDINATE	Z-COORDINATE	HCDE = 0	Z-DISPLACEMENT
1	9	G.CCCCO	1C.62C00	0.0	7.06320-04
2	9	C.36009	1C.6132C	0.0	2.53150-05
3	9	1.13234	10.55682	0.0	-1.52729-04
4	9	1.85674	10.45033	0.0	-2.55612-04
5	9	2.63346	1C.28831	0.0	-2.15757-03
6	9	3.CCCCO	10.18746	0.0	-3.32658-03
1	10	C.CCCCO	1C.7000C	0.0	3.03300-04
2	10	C.78CC2	1C.69325	0.0	3.87342-04
3	10	1.13813	10.6353C	0.0	-2.31415-04
4	10	1.69050	4.34901-04	0.0	-1.00560-03
5	10	2.63333	1C.3709C	0.0	-2.55236-03
6	10	3.CCCCO	10.27683	0.0	-4.27352-03
1	11	C.CCCCO	10.78600	0.0	5.73452-04
2	11	C.37454	1C.77330	0.0	7.17653-04
3	11	1.13763	10.71577	0.0	-2.33649-04
4	11	1.85026	10.61298	0.0	-1.12357-03
5	11	2.63321	1C.45345	0.0	-2.15065-03
6	11	3.CCCCO	1C.35415	0.0	-2.55715-03
1	12	C.CCCCO	10.5360C	0.0	5.74278-04
2	12	C.37580	1C.52340	0.0	4.27342-04
3	12	1.13756	10.87064	0.0	-2.45713-04
4	12	1.85583	1C.76338	0.0	-1.13277-03
5	12	2.63258	1C.60813	0.0	-2.15424-03
6	12	3.CCCCO	1C.51223	0.0	-2.55617-03
1	13	C.CCCCO	11.08C00	0.0	5.26860-04
2	13	C.37567	11.07349	0.0	3.77453-04
3	13	1.13721	11.62149	0.0	-2.69564-04
4	13	1.85542	10.91771	0.0	-1.14935-03
5	13	2.63275	10.76267	0.0	-2.16416-03
6	13	3.CCCCO	10.46613	0.0	-2.57356-03
1	14	C.CCCCO	11.16C00	0.0	5.16357-04
2	14	C.37560	11.15254	0.0	1.05742-04
3	14	1.13763	11.10193	0.0	-1.65381-04
4	14	1.85621	10.98093	0.0	-1.27264-03
5	14	2.63264	1C.84504	0.0	-7.31918-03
6	14	3.CCCCO	1C.74521	0.0	-1.15422-02
1	15	C.CCCCO	11.26C00	0.0	4.18000-04
2	15	C.37553	11.23359	0.0	2.58208-04
3	15	1.13686	11.18236	0.0	-7.12313-04
4	15	1.85650	3.21595-04	0.0	-1.07060-03
5	15	2.63253	1C.62737	0.0	-7.63527-03
6	15	3.CCCCO	1C.83225	0.0	-4.06217-02
1	16	C.CCCCO	11.32C00	0.0	2.51665-04
2	16	0.37546	11.31364	0.0	2.72960-04
3	16	1.13688	11.26276	0.0	-5.57912-04
4	16	1.85680	5.42693-04	0.0	-1.02496-03
5	16	2.63242	11.00867	0.0	-1.08950-02
6	16	3.CCCCO	10.91524	0.0	-2.55478-02

I	J	R	ENERGY DENSITY	AXIAL R	HOOP THETA	STRESSES AXIAL Z	STRAINS SHEAR R-Z	SHEAR R-T	SHEAR 2-T
1	1	0.273	9.248	-3.290430 01 2.790580-07	-3.307180 01 6.132320-08	-7.764710 C1 -5.788660 C5	1.003260-01 2.608470-07	0.0 0.0	0.0 0.0
2	1	0.865	9.239	-2.992560 01 1.234200-06	-3.078780 01 4.134230-07	-7.451270 C1 -5.642900-05	1.408070 C3 3.798270-06	0.0 0.0	0.0 0.0
3	1	1.504	9.219	-2.434560 01 2.795440-06	-2.572880 01 9.611890-07	-6.524880 01 -5.641480-05	4.548740 00 1.182670-05	0.0 0.0	0.0 0.0
4	1	2.141	9.187	-1.615820 01 3.316700-06	-1.768760 01 1.328500-06	-4.782490 C1 -3.785010-05	8.600710 00 2.236180-05	0.0 0.0	0.0 0.0
5	1	2.729	9.151	-9.402890 00 3.817270-06	-1.128090 01 1.375890-06	-3.260590 01 -2.634670-05	1.156280 01 3.006320-05	0.0 0.0	0.0 0.0
1	2	0.218	9.746	-3.425850 01 -2.621930-06	-3.380370 01 -7.868120-08	-7.605080 C1 -5.499590-05	-1.172670 00 -3.049930-06	0.0 0.0	0.0 0.0
2	2	0.796	9.721	-3.266140 01 -4.357430-08	-3.268170 01 -6.999150-08	-7.531770 C1 -5.544670-05	-2.262070 00 -5.861350-06	0.0 0.0	0.0 0.0
3	2	1.512	9.698	-2.486410 01 3.170370-06	-2.608890 01 9.021190-07	-6.701180 C1 -5.162170-05	-9.934260-01 -2.582510-06	0.0 0.0	0.0 0.0
4	2	2.223	9.554	-1.142610 01 7.625750-06	-1.553500 01 2.288110-08	-4.848650 C1 -4.054870-05	2.731240 00 7.101230-06	0.0 0.0	0.0 0.0
5	2	2.788	9.448	-1.224870 00 7.687220-06	-4.446560 00 3.498640-06	-2.448570 C1 -2.255190-05	7.594960 00 1.194690-05	0.0 0.0	0.0 0.0
1	3	0.190	10.036	-7.388120 01 -2.554820-04	-7.384650 01 -8.197330-05	-7.360220 C1 1.139390-03	-1.487470-01 -1.487460-03	0.0 0.0	0.0 0.0
2	3	0.760	10.004	-7.404860 01 2.824050-04	-7.412140 01 -7.555370-05	-7.419960 C1 -4.164130-04	3.483350-02 3.498320-04	0.0 0.0	0.0 0.0
3	3	1.516	9.518	-6.536700 01 2.757470-03	-6.991090 01 1.380030-04	-7.053000 C1 -2.957240-03	1.099290 00 1.097280-02	0.0 0.0	0.0 0.0
4	3	2.264	9.774	-4.426610 01 9.686870-03	-4.611610 01 4.389710-04	-4.825860 C1 -1.027430-02	3.200910 00 3.200990-02	0.0 0.0	0.0 0.0
5	3	2.817	9.635	-1.056810 01 1.705170-02	-1.355000 01 7.423350-04	-1.756200 C1 -1.781750-02	4.807850 00 4.807820-02	0.0 0.0	0.0 0.0
1	4	0.190	10.116	-7.346440 01 -4.571190-04	-7.345550 01 -4.126480-04	-7.333120 C1 2.089920-04	-3.561330-02 -3.561350-04	0.0 0.0	0.0 0.0

MODE = 0										
I	J	R	ENERGY DENSITY	Z	RACIAL R	HCOOP THETA	STRESSES /		SHEAR R-T	SHEAR Z-T
							AXIAL Z	SHEAR R-Z		
2	4	0.760	0.0	10.084	-7.446500 01 4.856430-04	-7.455500 01 2.056720-05	-7.458700 01 -1.244210-04	-3.665150-02 -3.665130-04	0.0 0.0	0.0 0.0
3	4	1.516	0.0	5.599	-6.885600 01 1.986420-03	-6.915630 01 8.143000-04	-6.977710 01 -2.518470-03	-1.312770-01 -1.312760-03	0.0 0.0	0.0 0.0
4	4	2.263	0.0	5.856	-7.656420 01 2.652900-03	-7.716430 01 1.253000-03	-7.815700 01 -3.215100-03	-3.709220-01 -3.709190-03	0.0 0.0	0.0 0.0
5	4	2.617	0.0	5.718	-1.336750 01 7.060500-03	-1.388530 01 1.571350-03	-1.534120 01 -5.707630-03	-1.739400-01 -1.708330-03	0.0 0.0	0.0 0.0
4	14	2.261	0.0	10.563	-7.552850 01 1.784140-02	-7.725020 01 1.732620-03	-5.292600 01 -1.914610-02	1.752650 00 1.752640-02	0.0 0.0	0.0 0.0
5	14	2.616	0.0	10.838	-1.822350 01 0.921400-02	-1.838910 01 5.928540-03	-4.700920 01 -7.496100-02	3.931280 00 3.931230-02	0.0 0.0	0.0 0.0
1	15	0.150	0.0	11.277	-4.565250 01 3.904600-04	-4.570500 01 2.776230-04	-4.545300 01 -5.393400-04	2.912460-02 2.912440-04	0.0 0.0	0.0 0.0
2	15	0.756	0.0	11.248	-4.577500 01 5.800200-04	-4.590420 01 4.571230-04	-5.008100 01 -5.287300-04	-4.925500-02 -4.905470-04	0.0 0.0	0.0 0.0
3	15	1.512	0.0	11.172	-7.566030 01 5.505600-05	-7.596310 01 3.004290-04	-4.573000 01 -4.552230-04	6.677350-02 6.677340-04	0.0 0.0	0.0 0.0
4	15	2.261	0.0	11.045	-7.456740 01 1.750410-02	-7.763230 01 2.660010-03	-9.211930 01 -1.575500-02	-5.076570-04 -5.076540-03	0.0 0.0	0.0 0.0
5	15	2.616	0.0	10.521	-1.362600 01 6.214700-02	-1.376500 01 7.054380-03	-7.225600 01 -6.590030-02	-3.510060 00 -3.510040-02	0.0 0.0	0.0 0.0





DISPLACEMENTS FOR ASYMMETRIC LOADING				MCSE = 1		Z-DISPLACEMENT
J	R-COORDINATE	Z-COORDINATE	R-DISPLACEMENT	TWETA-DISPLACEMENT		
1	0.0000	9.00000	0.0	0.0	0.0	0.0
1	C.6000	5.00000	0.0	0.0	0.0	C.C
1	1.2000	5.0000	C.0	0.0	0.0	C.C
1	1.8000	9.00000	0.0	0.0	0.0	C.C
1	2.4000	9.00000	0.0	0.0	0.0	C.0
1	3.0000	9.0000	0.0	0.0	0.0	C.0
1	C.6000	9.5000	0.0	0.0	0.0	C.0
2	C.4510	9.49203	-4.833770-05	4.826830-05	4.826830-05	-3.530850-05
2	1.17013	5.46451	-5.009260-05	4.831930-05	4.831930-05	1.362730-05
2	1.64635	5.41250	-5.964030-05	5.086590-05	5.086590-05	2.165230-05
2	2.51751	5.33625	-7.149990-05	5.021080-05	5.021080-05	1.658360-05
2	3.00000	5.26570	-8.083060-05	4.130190-06	4.130190-06	9.384390-06
2	C.6000	10.0000	-9.955150-05	2.715910-05	2.715910-05	4.453230-05
3	C.3877	9.59275	-1.145930-04	1.144930-04	1.144930-04	2.180080-07
3	1.14011	9.93479	-1.132700-04	1.127170-04	1.127170-04	2.445630-05
3	1.69284	5.81422	-1.117720-04	1.106550-04	1.106550-04	9.587360-05
3	2.63459	5.64671	-1.011020-04	1.011960-04	1.011960-04	4.001980-05
3	C.6000	5.53535	-1.533160-04	7.901260-05	7.901260-05	2.625680-05
3	C.6000	10.07281	-1.681330-04	5.670980-05	5.670980-05	9.570100-05
4	C.36063	10.07281	-3.192790-02	3.122740-02	3.122740-02	-2.740410-03
4	1.13586	10.01534	-3.190530-02	3.055270-02	3.055270-02	2.807440-03
4	1.65295	5.90074	-3.335230-02	3.191360-02	3.191360-02	2.603620-03
4	2.63443	5.72565	-3.700750-02	3.293870-02	3.293870-02	6.335420-03
4	3.0000	5.62322	-4.211170-02	3.435670-02	3.435670-02	1.071040-02
5	C.6000	10.16000	-4.950760-02	2.881100-02	2.881100-02	1.694400-02
5	C.38059	10.15287	-6.728040-02	6.729000-02	6.729000-02	-2.302640-03
5	1.13642	10.09586	-6.569140-02	6.492290-02	6.492290-02	3.944720-03
5	1.65226	9.98223	-6.799330-02	6.608800-02	6.608800-02	6.444130-03
5	2.63428	5.81255	-6.996970-02	6.660560-02	6.660560-02	1.502360-02
5	3.0000	9.70495	-7.593770-02	6.745730-02	6.745730-02	2.083730-02
5	C.6000	10.24000	-7.822840-02	9.167790-02	9.167790-02	2.799750-02
6	C.38050	10.23293	-1.033740-01	1.033720-01	1.033720-01	-1.262170-02
6	1.13938	10.17641	-1.032630-01	1.032730-01	1.032730-01	3.629790-03
6	1.65158	10.06370	-1.026430-01	1.025880-01	1.025880-01	1.183990-02
6	2.63413	5.89540	-1.014010-01	1.012600-01	1.012600-01	2.015330-02
6	3.0000	9.79065	-9.950220-02	9.930420-02	9.930420-02	2.865960-02
7	C.6000	10.39000	-1.048940-01	1.049000-01	1.049000-01	-0.576360-06
7	C.38033	10.38304	-1.048350-01	1.048240-01	1.048240-01	3.833580-03
7	1.13856	10.32733	-1.042880-01	1.041820-01	1.041820-01	3.833580-03
7	1.65148	10.21638	-1.031300-01	1.029080-01	1.029080-01	1.184590-02
7	2.63386	10.05062	-1.013040-01	1.010100-01	1.010100-01	2.616380-02
7	3.0000	9.54747	-1.010130-01	1.010100-01	1.010100-01	2.866740-02
8	C.6000	10.54000	-1.064230-01	1.064230-01	1.064230-01	3.254470-02
8	C.38017	10.53314	-1.064030-01	1.063700-01	1.063700-01	3.844560-03
8	1.13855	10.47833	-1.059260-01	1.057700-01	1.057700-01	1.186220-02
8	1.65099	10.36498	-1.045510-01	1.045500-01	1.045500-01	2.018660-02
8	2.63360	10.20567	-1.030990-01	1.027690-01	1.027690-01	2.667530-02
8	3.0000	10.10404	-1.019570-01	1.019570-01	1.019570-01	3.255500-02

DISPLACEMENTS FOR ASYMMETRIC LOADING

MODE = 1

I	J	R-COORDINATE	Z-COORDINATE	R-DISPLACEMENT	THETA-DISPLACEMENT	Z-DISPLACEMENT
1	9	C-00000	10.62000	-1.24549D-01	1.26170C-01	-1.65275D-03
2	9	C-30009	10.61320	-1.24112D-01	1.23324C-01	5.37546D-03
3	9	1.13834	10.59882	-1.23904D-01	1.23302D-01	1.30009D-02
4	9	1.85074	10.45033	-1.25703D-01	1.22807C-01	2.46126D-02
5	9	2.63346	10.28831	-1.30310D-01	1.22265D-01	3.52546C-02
6	9	C-00000	10.18742	-1.33831D-01	1.18709C-01	4.39285D-02
1	10	C-00000	10.70000	-1.44478D-01	1.44184D-01	-1.11833D-03
2	10	C-30002	10.69325	-1.43168D-01	1.42706C-01	5.72083C-03
3	10	1.13813	10.63936	-1.43318D-01	1.42552D-01	1.52463D-02
4	10	1.85050	10.53167	-1.44191D-01	1.41769D-01	2.81612C-02
5	10	2.63233	10.37090	-1.48851D-01	1.46904D-01	4.10360D-02
6	10	C-00000	10.27083	-1.52129D-01	1.37268D-01	5.00324D-02
1	11	C-00000	10.78000	-1.64712D-01	1.64708D-01	-5.60831D-06
2	11	C-37544	10.77330	-1.64497D-01	1.64536D-01	5.63265D-03
3	11	1.13793	10.71577	-1.63493D-01	1.63489D-01	1.74138D-02
4	11	1.85026	10.61298	-1.61531D-01	1.61521D-01	3.08715D-02
5	11	2.63221	10.45345	-1.58687D-01	1.58677D-01	4.44373D-02
6	11	C-00000	10.35415	-1.56844D-01	1.56937D-01	5.12514D-02
1	12	C-00000	10.93000	-1.66941D-01	1.66942C-01	-1.65432D-05
2	12	C-37580	10.92340	-1.66848D-01	1.66827D-01	5.62851D-03
3	12	1.13756	10.87064	-1.66031D-01	1.65962D-01	1.78131D-02
4	12	1.85083	10.76536	-1.64278D-01	1.64032C-01	3.08732D-02
5	12	2.63258	10.60613	-1.61513D-01	1.61298D-01	4.44352D-02
6	12	C-00000	10.51023	-1.59737D-01	1.59605D-01	5.12460D-02
1	13	C-00000	11.08000	-1.69168D-01	1.69175C-01	-5.00600C-06
2	13	C-37567	11.07345	-1.69188D-01	1.69113D-01	5.64070D-03
3	13	1.13721	11.02149	-1.68555D-01	1.68267D-01	1.78326D-02
4	13	1.85042	10.91771	-1.66987D-01	1.66533C-01	3.08682D-02
5	13	2.63275	10.76267	-1.64349D-01	1.63909D-01	4.44510D-02
6	13	C-00000	10.66613	-1.62630D-01	1.62261D-01	5.12650D-02
1	14	C-00000	11.16000	-1.70412D-01	1.70371C-01	-5.05502D-04
2	14	C-37560	11.15354	-1.70362D-01	1.70370D-01	5.10528D-03
3	14	1.13703	11.10193	-1.70136D-01	1.69580C-01	1.76210D-02
4	14	1.85021	10.99893	-1.68233D-01	1.68087C-01	3.11014D-02
5	14	2.63264	10.84504	-1.79430D-01	1.68777D-01	4.87473D-02
6	14	C-00000	10.74921	-2.06893D-01	1.64537D-01	5.05950D-02
1	15	C-00000	11.24000	-1.71786D-01	1.71775D-01	5.47316D-05
2	15	C-37953	11.23354	-1.71586D-01	1.71617D-01	5.65247D-03
3	15	1.13686	11.18236	-1.71566D-01	1.70384D-01	1.61360D-02
4	15	1.85000	11.08013	-1.69571D-01	1.69352C-01	3.05520D-02
5	15	2.63253	10.92737	-1.89681D-01	1.68238D-01	5.53349D-02
6	15	C-00000	10.83225	-2.27519D-01	1.66353D-01	2.71175D-02
1	16	C-00000	11.32000	-1.72927D-01	1.72963D-01	5.67922D-05
2	16	C-37546	11.31364	-1.72690D-01	1.72357D-01	5.61260D-03
3	16	1.13668	11.26279	-1.72800D-01	1.72176C-01	1.63649D-02
4	16	1.85080	11.16131	-1.71247D-01	1.70920C-01	3.05220D-02
5	16	2.63242	11.00967	-1.93780D-01	1.71223C-01	6.17840D-02
6	16	C-00000	10.91524	-2.34687D-01	1.69170C-01	1.03592D-01

I J		R		ENERGY DENSITY		RACIAL R		HCOOP THETA		STRESSES		STRAINS		SHEAR		SHEAR	
										AXIAL Z		SHEAR R-Z		SHEAR R-T		SHEAR Z-T	
1 1		0.273		0.0		5.526440 OC		7.161170 00		2.095530 C1		-3.321700 01		E.411760-01		3.632130 C1	
2 1		0.865		0.0		-2.713990-06		-1.111440-06		1.682100-C5		-8.636620-05		2.187000-06		9.443530-05	
3 1		1.504		0.0		7.257360 OC		1.317940 01		9.348010 C1		-4.174280 01		2.347340 00		3.685240 01	
4 1		2.141		0.0		-9.726580-06		-2.027950-06		3.736300-05		-1.085310-04		6.103000-06		9.531630-05	
5 1		2.725		0.0		1.372310 OC		1.020600 01		4.606680 01		-5.800100 01		3.013320 00		7.181770 01	
6 1		0.216		0.0		-1.525160-05		-3.710840-06		4.377330-C5		-1.508020-04		7.634630-06		1.037260-C4	
7 1		0.796		0.0		-1.363850 01		-2.725860 00		2.449150 01		-7.948940 01		2.202580 00		4.565480 C1	
8 1		0.216		0.0		-1.934230-05		-5.355630-06		3.002680-C5		-2.066720-C4		5.727450-06		1.192220-C4	
9 1		0.796		0.0		-1.468570 01		1.271340 01		6.638150 C1		-9.793120 01		-2.618550-01		4.294250 C1	
10 1		0.216		0.0		-4.202550-05		-6.406630-06		8.938190-C5		-2.546210-04		-6.808230-07		1.116510-C4	
11 2		0.796		0.0		5.141120 OC		4.227210 00		2.096540 C1		-3.259720 01		1.708650 00		3.722020 01	
12 2		0.796		0.0		-2.750130-06		-3.978210-06		1.778140-C5		-8.472670-05		4.442480-06		9.677260-C5	
13 2		0.796		0.0		6.975230 00		1.256820 01		4.994190 C1		-3.663380 01		5.190250 00		3.762610 01	
14 2		0.796		0.0		-1.282420-05		-5.553270-06		4.303250-C5		-9.537790-05		1.349470-05		9.762780-05	
15 2		0.796		0.0		-9.022740 00		6.456490 00		5.914190 C1		-4.570570 01		6.113500 00		4.038690 01	
16 2		0.796		0.0		-2.631500-05		-8.152100-06		5.899910-C5		-1.108350-04		1.589610-C5		1.050110-C4	
17 2		0.796		0.0		-5.011010 01		-1.653030 01		4.441680 C1		-6.307030 01		5.561030 00		4.403610 01	
18 2		0.796		0.0		-5.782230-05		-1.468830-05		6.506260-C5		-1.639830-04		1.445870-05		1.144540-C4	
19 2		0.796		0.0		-9.814580 01		-4.814570 01		2.302350 C1		-4.201700 01		1.952740 00		4.090070 C1	
20 2		0.796		0.0		-9.620450-05		-2.520470-05		6.731500-C5		-1.092440-C4		5.077120-06		1.063420-C4	
21 3		0.190		0.0		1.815440 01		1.887690 01		1.671120 C1		-3.900660 01		1.029580 00		3.909110 C1	
22 3		0.760		0.0		-7.506120-03		-3.893940-03		2.774260-C4		-3.900660-01		1.029580-02		3.909090-C1	
23 3		0.760		0.0		3.365100 01		3.949440 01		4.664380 C1		-4.025880 01		3.187210 00		3.857780 C1	
24 3		0.760		0.0		-3.172420-02		-1.516210-03		3.322940-C2		-4.025880-01		3.187190-02		3.857760-01	
25 3		0.760		0.0		4.786140 01		6.145630 01		7.502910 C1		-4.189400 01		6.245280 00		3.979630 C1	
26 3		0.760		0.0		-6.863800-02		-7.640530-04		6.709920-C2		-4.189370-01		6.245240-02		3.979610-C1	
27 3		0.760		0.0		-1.155380 01		1.075720 01		3.406870 C1		-4.514790 01		9.654870 00		4.064880 C1	
28 3		0.760		0.0		-1.145860-01		-1.031980-03		1.159250-C1		-4.514760-01		9.654810-02		4.064850-01	
29 3		0.760		0.0		-6.935510 01		-3.752660 01		-4.716700 C0		-4.740580 01		1.044840 01		3.761170 C1	
30 3		0.760		0.0		-1.606720-01		-1.529370-03		1.623160-C1		-4.740540-01		1.044840-01		3.761140-C1	
31 4		0.190		0.0		1.055690 01		1.089470 01		1.412900 C1		-4.162770 01		1.127460 00		4.347550 C1	
32 4		0.190		0.0		-6.006450-03		-6.320750-03		4.490850-C3		-4.162750-01		1.127450-02		4.347520-C1	

MODE = 1									
I	J	R	ENERGY DENSITY	RADIAL R	MCOP TMEYA	STRESSES / AXIAL Z	STRAINS SHEAR R-Z	SHEAR R-T	SHEAR Z-T
2	4	0.760	0.0	10.064	4.347540 01 -2.120860-03	5.060590 C1 3.353130-C2	-4.218310 01 -4.218480-01	3.596360 00 3.596330-C2	4.222670 C1 4.222240-C1
3	4	1.316	0.0	9.699	4.205150 01 -6.703270-02	6.901270 C1 6.757230-C2	-3.979150 01 -3.979120-01	6.662460 00 6.662420-C2	4.147400 C1 4.137380-C1
4	4	2.263	0.0	9.856	-1.240680 00 -1.088230-01	3.941470 01 1.024520-C1	-3.775130 01 -3.775110-01	9.734500 00 9.734430-C2	4.006040 C1 4.006020-C1
5	4	2.817	0.0	9.718	-7.554170 01 -1.231490-01	-2.541610 C1 1.274280-C1	-3.393320 01 -3.393300-01	1.038600 01 1.038560-C1	3.842660 C1 3.842630-C1
1	15	0.190	0.0	11.277	-1.227050-01 7.576070-05	-1.297250-C1 4.068170-C5	-8.628460-02 -8.628350-04	-3.188880-C2 -3.188840-04	5.187800-C3 5.187830-C5
2	15	0.758	0.0	11.248	-2.520580-01 -1.021410-03	1.856910-C1 1.167520-C3	8.965690-02 8.965640-04	5.625750-C2 5.625750-04	8.651230-C3 8.651170-C5
3	15	1.313	0.0	11.172	-5.587640-01 -8.065450-05	-4.707090-C1 5.595750-C5	-1.528370-01 -1.528360-03	7.406560-02 7.406530-04	2.764660-C2 2.764640-C4
4	15	2.261	0.0	11.045	-1.047810 01 -3.594880-02	3.275390 C1 3.882610-C2	1.006000 00 1.007960-02	8.780730-01 8.780670-C3	1.387080-C1 1.387080-C3
5	15	2.816	0.0	10.921	-6.205720 01 -1.245260-01	-9.495240 C1 1.382870-C1	7.037540 00 7.007490-02	1.547760 00 1.547650-02	-5.344130-C4 -5.344090-C4

MODEL PT	COORDINATES	DISPLACEMENTS	U	V	W
1	0.0000	9.0000	0.0	0.0	0.0
2	0.0000	9.0000	0.0	0.0	0.0
3	1.2000	9.0000	0.0	0.0	0.0
4	1.8000	9.0000	0.0	0.0	0.0
5	2.4000	9.0000	0.0	0.0	0.0
6	3.0000	9.0000	0.0	0.0	0.0
7	0.0000	9.0000	0.0	0.0	0.0
8	0.0000	9.0000	0.0	0.0	0.0
9	0.0000	9.0000	0.0	0.0	0.0
10	0.0000	9.0000	0.0	0.0	0.0
11	0.0000	9.0000	0.0	0.0	0.0
12	0.0000	9.0000	0.0	0.0	0.0
13	0.0000	9.0000	0.0	0.0	0.0
14	0.0000	9.0000	0.0	0.0	0.0
15	0.0000	9.0000	0.0	0.0	0.0
16	0.0000	9.0000	0.0	0.0	0.0
17	0.0000	9.0000	0.0	0.0	0.0
18	0.0000	9.0000	0.0	0.0	0.0
19	0.0000	9.0000	0.0	0.0	0.0
20	0.0000	9.0000	0.0	0.0	0.0
21	0.0000	9.0000	0.0	0.0	0.0
22	0.0000	9.0000	0.0	0.0	0.0
23	0.0000	9.0000	0.0	0.0	0.0
24	0.0000	9.0000	0.0	0.0	0.0
25	0.0000	9.0000	0.0	0.0	0.0
26	0.0000	9.0000	0.0	0.0	0.0
27	0.0000	9.0000	0.0	0.0	0.0
28	0.0000	9.0000	0.0	0.0	0.0
29	0.0000	9.0000	0.0	0.0	0.0
30	0.0000	9.0000	0.0	0.0	0.0
31	0.0000	9.0000	0.0	0.0	0.0
32	0.0000	9.0000	0.0	0.0	0.0
33	0.0000	9.0000	0.0	0.0	0.0
34	0.0000	9.0000	0.0	0.0	0.0
35	0.0000	9.0000	0.0	0.0	0.0
36	0.0000	9.0000	0.0	0.0	0.0
37	0.0000	9.0000	0.0	0.0	0.0
38	0.0000	9.0000	0.0	0.0	0.0
39	0.0000	9.0000	0.0	0.0	0.0
40	0.0000	9.0000	0.0	0.0	0.0
41	0.0000	9.0000	0.0	0.0	0.0
42	0.0000	9.0000	0.0	0.0	0.0
43	0.0000	9.0000	0.0	0.0	0.0
44	0.0000	9.0000	0.0	0.0	0.0
45	0.0000	9.0000	0.0	0.0	0.0
46	0.0000	9.0000	0.0	0.0	0.0
47	0.0000	9.0000	0.0	0.0	0.0
48	0.0000	9.0000	0.0	0.0	0.0
49	0.0000	9.0000	0.0	0.0	0.0
50	0.0000	9.0000	0.0	0.0	0.0
51	0.0000	9.0000	0.0	0.0	0.0
52	0.0000	9.0000	0.0	0.0	0.0
53	0.0000	9.0000	0.0	0.0	0.0
54	0.0000	9.0000	0.0	0.0	0.0
55	0.0000	9.0000	0.0	0.0	0.0
56	0.0000	9.0000	0.0	0.0	0.0
57	0.0000	9.0000	0.0	0.0	0.0
58	0.0000	9.0000	0.0	0.0	0.0
59	0.0000	9.0000	0.0	0.0	0.0
60	0.0000	9.0000	0.0	0.0	0.0
61	0.0000	9.0000	0.0	0.0	0.0
62	0.0000	9.0000	0.0	0.0	0.0
63	0.0000	9.0000	0.0	0.0	0.0
64	0.0000	9.0000	0.0	0.0	0.0
65	0.0000	9.0000	0.0	0.0	0.0
66	0.0000	9.0000	0.0	0.0	0.0
67	0.0000	9.0000	0.0	0.0	0.0
68	0.0000	9.0000	0.0	0.0	0.0
69	0.0000	9.0000	0.0	0.0	0.0
70	0.0000	9.0000	0.0	0.0	0.0
71	0.0000	9.0000	0.0	0.0	0.0
72	0.0000	9.0000	0.0	0.0	0.0
73	0.0000	9.0000	0.0	0.0	0.0
74	0.0000	9.0000	0.0	0.0	0.0
75	0.0000	9.0000	0.0	0.0	0.0
76	0.0000	9.0000	0.0	0.0	0.0
77	0.0000	9.0000	0.0	0.0	0.0
78	0.0000	9.0000	0.0	0.0	0.0
79	0.0000	9.0000	0.0	0.0	0.0
80	0.0000	9.0000	0.0	0.0	0.0
81	0.0000	9.0000	0.0	0.0	0.0
82	0.0000	9.0000	0.0	0.0	0.0
83	0.0000	9.0000	0.0	0.0	0.0
84	0.0000	9.0000	0.0	0.0	0.0
85	0.0000	9.0000	0.0	0.0	0.0
86	0.0000	9.0000	0.0	0.0	0.0
87	0.0000	9.0000	0.0	0.0	0.0
88	0.0000	9.0000	0.0	0.0	0.0
89	0.0000	9.0000	0.0	0.0	0.0
90	0.0000	9.0000	0.0	0.0	0.0
91	0.0000	9.0000	0.0	0.0	0.0
92	0.0000	9.0000	0.0	0.0	0.0
93	0.0000	9.0000	0.0	0.0	0.0
94	0.0000	9.0000	0.0	0.0	0.0
95	0.0000	9.0000	0.0	0.0	0.0
96	0.0000	9.0000	0.0	0.0	0.0
97	0.0000	9.0000	0.0	0.0	0.0
98	0.0000	9.0000	0.0	0.0	0.0
99	0.0000	9.0000	0.0	0.0	0.0
100	0.0000	9.0000	0.0	0.0	0.0

COORDINATES AND DISPLACEMENTS FOR THETA = 0.0

NODAL PT	COORDINATES		DISPLACEMENTS		M
	U	V	U	V	
1 9	0.0000	10.6200	-1.24545E-01	0.0	-1.24637E-03
2 9	0.3801	10.6132	-1.24608E-01	0.0	5.40097E-03
3 9	1.1383	10.5908	-1.24660E-01	0.0	1.28481E-02
4 9	1.8507	10.4503	-1.25197E-01	0.0	2.37573E-02
5 9	2.6335	10.2883	-1.25918E-01	0.0	3.30572E-02
6 9	3.0000	10.1875	-1.27115E-01	0.0	4.01503E-02
1 10	0.0000	10.7000	-1.44476E-01	0.0	-8.15926E-04
2 10	0.3800	10.6932	-1.43687E-01	0.0	6.10742E-03
3 10	1.1381	10.6393	-1.44165E-01	0.0	1.50149E-02
4 10	1.8505	10.5317	-1.43757E-01	0.0	2.71556E-02
5 10	2.6333	10.3709	-1.45038E-01	0.0	3.81636E-02
6 10	3.0000	10.2708	-1.44938E-01	0.0	4.57584E-02
1 11	0.0000	10.7800	-1.64712E-01	0.0	5.07883E-04
2 11	0.3755	10.7723	-1.64550E-01	0.0	6.05033E-03
3 11	1.1375	10.7158	-1.63555E-01	0.0	1.75674E-02
4 11	1.8503	10.6130	-1.61691E-01	0.0	2.97404E-02
5 11	2.6332	10.4525	-1.58962E-01	0.0	4.22645E-02
6 11	3.0000	10.3541	-1.57254E-01	0.0	4.85243E-02
1 12	0.0000	10.9300	-1.60941E-01	0.0	5.57335E-04
2 12	0.3756	10.9234	-1.60802E-01	0.0	6.03585E-03
3 12	1.1376	10.8706	-1.65542E-01	0.0	1.75674E-02
4 12	1.8506	10.7654	-1.64226E-01	0.0	2.97404E-02
5 12	2.6330	10.6081	-1.61600E-01	0.0	4.22645E-02
6 12	3.0000	10.5102	-1.59937E-01	0.0	4.85243E-02
1 13	0.0000	11.0800	-1.69168E-01	0.0	5.22855E-04
2 13	0.3757	11.0735	-1.69045E-01	0.0	6.02470E-03
3 13	1.1372	11.0215	-1.68317E-01	0.0	1.75636E-02
4 13	1.8504	10.9177	-1.66749E-01	0.0	2.97395E-02
5 13	2.6322	10.7627	-1.64229E-01	0.0	4.22865E-02
6 13	3.0000	10.6661	-1.62615E-01	0.0	4.85865E-02
1 14	0.0000	11.1600	-1.70412E-01	0.0	1.10654E-04
2 14	0.3756	11.1535	-1.70262E-01	0.0	6.21502E-03
3 14	1.1370	11.1019	-1.69712E-01	0.0	1.74007E-02
4 14	1.8502	10.9909	-1.67974E-01	0.0	2.98288E-02
5 14	2.6326	10.8450	-1.72402E-01	0.0	4.44292E-02
6 14	3.0000	10.7452	-1.65395E-01	0.0	5.74434E-02
1 15	0.0000	11.2400	-1.71786E-01	0.0	4.72731E-04
2 15	0.3755	11.2336	-1.71404E-01	0.0	5.95046E-03
3 15	1.1369	11.1824	-1.71033E-01	0.0	1.77047E-02
4 15	1.8500	11.0801	-1.65255E-01	0.0	2.55683E-02
5 15	2.6325	10.9274	-1.70165E-01	0.0	4.76541E-02
6 15	3.0000	10.8322	-1.66568E-01	0.0	6.44562E-02
1 16	0.0000	11.3200	-1.72927E-01	0.0	3.91457E-04
2 16	0.3755	11.3136	-1.72726E-01	0.0	5.86096E-03
3 16	1.1367	11.2628	-1.72249E-01	0.0	1.78025E-02
4 16	1.8500	11.1613	-1.70704E-01	0.0	2.94570E-02
5 16	2.6324	11.0057	-1.60463E-01	0.0	5.06835E-02
6 16	3.0000	10.9152	-2.60594E-01	0.0	7.46740E-02

R		S T R E S S / S T R A I N S		R-TMETHA		Z-TMETHA		R-Z	
R		T M E T H A		P L E V A L U E S		P L E V A L U E S		P L E V A L U E S	
1	1	STRESS	-0.2697583C C2	-0.5669173D 02	0.0	0.0	0.0	-0.3311745D 02	
0.27	9.25	STRAIN	-0.2436533D-C3	-0.1650118D-05	0.0	0.0	0.0	-0.0610537D-04	
M=	-0.1687780C-C3								
1	1	STRESS	-0.5536665D C1	-0.7813150D 02	0.1018726D 02	0.362772D 02	0.2611045D 02		
		STRAIN	0.2543676D-C4	-0.1050118D-C5	0.2640688D-04	0.9437406D-04	0.6768718D-C4		
2	1	STRESS	-0.2266627D C2	-0.176C839D 02	0.0	0.0	0.0	-0.4028191D 02	
0.07	9.24	STRAIN	-0.0192382C-05	-0.1614530D-05	0.0	0.0	0.0	-0.1047330D-03	
M=	-0.7092126D-C4								
1	1	STRESS	0.13646C0D C2	-0.176C839D 02	0.156282C0 C2	0.4049842D 02	0.2487023D 02		
		STRAIN	0.3901878C-04	-0.1614530D-C5	0.4063331D-C4	0.1052959D-03	0.6466259D-04		
3	1	STRESS	-0.2267325D C2	-0.1544825D C2	0.0	0.0	0.0	-0.5345221C 02	
1.50	9.22	STRAIN	-0.1253214D-C4	-0.2745654D-05	0.0	0.0	0.0	-0.1389757C-03	
M=	-0.5777652D-C4								
1	1	STRESS	0.3275298D C2	-0.1544825D 02	0.2412642D C2	0.5350020D C2	0.2537978D 02		
		STRAIN	0.5996345D-C4	-0.2745654D-05	0.6271310D-C4	0.1391005C-03	0.7638742D-C4		
4	1	STRESS	-0.2976665D C2	-0.2641341D C2	0.0	0.0	0.0	-0.7088868C C2	
2.14	9.19	STRAIN	-0.162256C0-04	-0.4627335D-05	0.0	0.0	0.0	-0.1843106D-03	
M=	-0.7503386D-04								
1	1	STRESS	0.4439729C C2	-0.2641341D 02	0.324C535D C2	0.7096230D 02	0.3855696D 02		
		STRAIN	0.8022657D-04	-0.4627335D-05	0.4425591D-C4	0.1845020D-03	0.1002481C-03		
5	1	STRESS	-0.2408855C C2	0.1432526D C1	0.0	0.0	0.0	-0.0636840C 02	
2.73	9.15	STRAIN	-0.3820119D-C4	-0.503C734D-C5	0.0301522D-04	0.0	0.0	-0.2245579D-03	
M=	0.2297673D-04								
1	1	STRESS	0.1055611C C3	0.1432526D C1	0.5407427D C2	0.9473757D 02	0.4066331C C2		
		STRAIN	0.1355624C-C3	-0.503C734D-05	0.1107553D-03	0.2463177D-03	0.1C57246D-03		
1	2	STRESS	-0.3015735C C2	-0.2957646D 02	0.0	0.0	0.0	-0.3373987C 02	
0.22	9.75	STRAIN	-0.4012C56C-05	-0.4036891D-05	0.0	0.0	0.0	-0.0777567D-04	
M=	-0.1146417D-03								
1	1	STRESS	-0.6634143C C1	-0.2557646D 02	0.1147116D C2	0.3598723D 02	0.2451607C 02		
		STRAIN	0.2576612D-04	-0.4036891D-05	0.2982561D-04	0.9356680D-04	0.6374179D-04		
2	2	STRESS	-0.2566615D C2	-0.2641347D C2	0.0	0.0	0.0	-0.3894587D 02	
0.00	9.72	STRAIN	-0.1286775D-04	-0.5623265D-05	0.0	0.0	0.0	-0.1012553D-03	
M=	-0.6441417D-04								
1	1	STRESS	0.1341522D C2	-0.2641347D C2	0.1676434D 02	0.3894618D 02	0.2218184D C2		
		STRAIN	0.37964C3D-C4	-0.5623265D-05	0.4355673D-C4	0.1012601C-03	0.5767278C-04		

PROG 53399 SAMPLE PROBLEM 2. DATE 8/20/75 TIME 08:57 PAGE 24

FIRST HARMONIC

GLOBAL PT COORDINATES

3 2 STRESS -0.33081800 C2  
1.51 9.66 STRAIN -0.25144600 C4  
M= -0.63027448 C4  
C.26966480 C2  
0.53965280 C4

R		S T R E S S / S T R A I N S		R-THETA		Z-THETA		R-Z	
		T H E T A							
3	2	STRESS	-0.20152470 C2	-0.88098890 C1	0.0	0.0	0.0	-0.70099140 C2	
	9.66	STRAIN	-0.72859820 C5	0.73773780 C5	0.0	0.0	0.0	-0.12141770 C3	
	M=		P R I M C I P L E V A L U E S						
			C.26966480 C2		0.23555720 C2		0.7342300 C2		0.24785550 C2
4	2	STRESS	-0.32405310 C2	-0.40090150 C1	0.0	0.0	0.0	-0.63335040 C2	
	9.66	STRAIN	-0.12400350 C4	0.24514050 C4	0.0	0.0	0.0	-0.15080150 C3	
	M=		P R I M C I P L E V A L U E S						
			C.34028220 C2		0.33240770 C2		0.60031150 C2		0.33534300 C2
5	2	STRESS	-0.12400350 C4	-0.59719740 C4	0.0	0.0	0.0	-0.37319350 C4	
	9.66	STRAIN	-0.12400350 C4	-0.59719740 C4	0.0	0.0	0.0	-0.37319350 C4	
	M=		P R I M C I P L E V A L U E S						
			C.34028220 C2		0.33240770 C2		0.60031150 C2		0.33534300 C2

4	15	STRESS	-0.52041250 C2	-0.47043890 C2	0.0	0.0	0.0	-0.30034150 C3	
	11.04	STRAIN	-0.19153400 C2	0.19071150 C1	0.0	0.0	0.0	-0.50033050 C2	
	M=		P R I M C I P L E V A L U E S						
			C.26966480 C2		0.23555720 C2		0.7342300 C2		0.24785550 C2
5	15	STRESS	-0.65050510 C2	-0.54751710 C2	0.0	0.0	0.0	-0.34574750 C1	
	10.52	STRAIN	-0.62373400 C1	0.68787100 C1	0.0	0.0	0.0	-0.34574750 C1	
	M=		P R I M C I P L E V A L U E S						
			C.34028220 C2		0.33240770 C2		0.60031150 C2		0.33534300 C2

PLOT FOR 1 1 6 WITH X & Y SCALE FACTORS 2.000000 C0 2.000000 C0  
THE X & Y MINS & MAXS ARE 8.000000 C0 -1.000000 C0 1.903795D-77 1.903795D-77



# THICKOL AUTOMATED STRESS SYSTEM

```

10:58 ENTERED INPUT MCCLLE
      INPUT ROUTINE REQUIRED      3K BYTES. ADDITIONAL STORAGE MAY BE REQUIRED BY WORKING ROUTINES.
      INPUT MCCLLE HAD 62 K AVAILABLE AND USED 6 K. EXCESS AVAILABLE WAS 56 K.
10:59 LEAVING INPUT MCCLLE. CPU TIME 0.085 MINS., WAIT TIME 0.054 MINS.

10:59 ENTERED SOLUTION PCCLLE
      SOLUTION MODULE HAD 62 K AVAILABLE AND USED 13 K. EXCESS AVAILABLE WAS 49 K.
11:01 LEAVING SOLUTION MCCLLE. CPU TIME 1.146 MINS., WAIT TIME 0.058 MINS.

11:01 ENTERED STRESS PCCLLE
      STRESS MODULE HAD 62 K AVAILABLE AND USED 2 K. EXCESS AVAILABLE WAS 61 K.
11:01 LEAVING STRESS MCCLLE. CPU TIME 1.194 MINS., WAIT TIME 0.061 MINS.

11:01 ENTERED PRINT MCCLLE
      PRINT MCCLLE HAD 62 K AVAILABLE AND USED 1 K. EXCESS AVAILABLE WAS 61 K.
11:01 LEAVING PRINT MCCLLE. CPU TIME 1.220 MINS., WAIT TIME 0.064 MINS.

```

# THICKOL AUTOMATED STRESS SYSTEM

```

11:01 ENTERED INPUT MCCLLE
      INPUT ROUTINE REQUIRED      3K BYTES. ADDITIONAL STORAGE MAY BE REQUIRED BY WORKING ROUTINES.
      INPUT MCCLLE HAD 62 K AVAILABLE AND USED 7 K. EXCESS AVAILABLE WAS 55 K.
11:02 LEAVING INPUT MCCLLE. CPU TIME 1.285 MINS., WAIT TIME 0.074 MINS.

11:02 ENTERED SOLUTION PCCLLE
      SOLUTION MODULE HAD 62 K AVAILABLE AND USED 13 K. EXCESS AVAILABLE WAS 49 K.
11:04 LEAVING SOLUTION MCCLLE. CPU TIME 2.504 MINS., WAIT TIME 0.077 MINS.

11:04 ENTERED STRESS PCCLLE
      STRESS MODULE HAD 62 K AVAILABLE AND USED 2 K. EXCESS AVAILABLE WAS 61 K.
11:04 LEAVING STRESS MCCLLE. CPU TIME 2.552 MINS., WAIT TIME 0.080 MINS.

11:04 ENTERED PRINT MCCLLE
      PRINT MCCLLE HAD 62 K AVAILABLE AND USED 1 K. EXCESS AVAILABLE WAS 61 K.
11:04 LEAVING PRINT MCCLLE. CPU TIME 2.578 MINS., WAIT TIME 0.082 MINS.

11:04 ENTERED PLOT/ACCUMULATION MODULE
      P/A MODULE HAD 62 K BYTES AVAILABLE AND USED 4 K BYTES. EXCESS AVAILABLE WAS 56 K BYTES.
11:05 LEAVING PLOT/ACCUMULATION MODULE. CPU TIME 2.634 MINS., WAIT TIME 0.089 MINS.

```

## APPENDIX A

### Service Life Program

This program is designated as S3359SL. It is designed to accept a tape or tapes created by S3359, when output option 20 has been specified, and accumulate a selected set of stresses and strains.

The energy is calculated for each element from the principal stresses and strains as the accumulation is done.

The accumulated energy, stresses and strains and the principal stresses and strains calculated from the accumulated stresses and strains are printed out at the end of the run. The locations of the highest energy and the highest value of the principal stresses are printed out also.

#### Input

When the S3359 runs were made, a pair of integers giving the date and time for each set of data put on the tape was written out. To select the desired sets of data from the tape for accumulation, the date and time of each of the desired sets must be input exactly as they were written out and in the same order as they appear on the tape.

The input is via FREFRM as follows:

Title - A title record that will be printed on the top of each page of the output. This must be only one card.

#### Control Records

As many records as desired to select the data sets.

- |    |  |
|----|--|
| L1 | The data as printed by S3359.  |
| L2 | The time as printed by S3359.  |
| L3 | The factor by which the stresses and strains will be multiplied. The default value is 1.0. |

An end of file stops the reading and accumulating and causes the final output to be produced.

## APPENDIX B

### S3359F Fourier Coefficient Generator

This program computes the Fourier series coefficients by use of Trapezoidal Integration Formula, then fits the function  $f(\theta)$  with a Fourier series curve fit. The output includes a set of cards containing the Fourier series coefficients in a form compatible with the input requirements of program S3359.

The function to be fit with a Fourier series curve fit must be either odd or even having period  $2\pi$ . The integration for the coefficients is performed over the interval  $[0, \pi]$  and then multiplied by two.

#### Preparation of Input

The input is read in in free form. (The rules of free-form input are given in the S3359 document in detail.) The first card is a title card. The second card contains data specifying the symmetry of the function, the error tolerance which is compared with the least-squares error approximation, the maximum harmonic which is to be calculated, and a plot flag specifying what is to be plotted. The next set of cards contains values of  $\theta$  and  $f(\theta)$  where  $\theta$  is in degrees.

#### Title Record

May contain any desired alphanumeric information

#### Control Record

IOEF	IOEF=0 if $f(\theta)$ is even, 1 if $f(\theta)$ is odd
ISYM	Specifies the symmetry of $f(\theta)$ as shown in Tables B-1 and B-2
E	Real number error tolerance allowed for Fourier series curve fit
N	Integer number specifying the maximum harmonic to be computed. Maximum of fifteen.
IPLLOT	IPLLOT=0, no plot

IPL0T=1, plot all harmonics

IPL0T=2, plot final harmonic only

IPL0T=3, plot specified harmonics

IH(15)                      Integer number specifying harmonic numbers to be used in plotting. Maximum of fifteen. Required only for IPL0T=3.

#### Input Data Cards

$\theta$                       Real value of  $\theta$  in degrees

$f(\theta)$                   Real value of  $f(\theta)$

The title card must be ended with a semicolon. All other data is separated by commas and ended with a semicolon. In choosing N and IH, the user must keep in mind what type of symmetry (IOEF and ISYM) he is working with. For example, if IOEF = 0, and ISYM = 1, only even harmonics are calculated since the odd harmonics are zero. Hence, N and the values of IH should be even.

If the Fourier coefficients are to be computed for more than one function, the data for each function is preceded by a title record.

Many functions of interest contain some type of symmetry. In these cases certain terms in the Fourier series will be zero and therefore are not calculated. The function  $f$ , in this case, is always either odd or even. The following rules of symmetry apply.

EVEN FUNCTION  
Table B-1

<u>Code</u>	<u>Function</u>	<u>Fourier Terms</u>
ISYM=0	$f(\theta) = f(-\theta)$	no sine terms
ISYM=1 or 2	$f(\theta) = f(180+\theta) = f(180-\theta)$	no sine terms; no odd cosine terms
ISYM=3	$f(\theta) - f(180+\theta) = -f(180-\theta)$	no constant term  no even cosine terms  no sine terms

ODD FUNCTION  
Table B-2

<u>Code</u>	<u>Function</u>	<u>Fourier Terms</u>
ISYM=0	$f(-\theta) = -f(\theta)$	no constant term  no cosine terms
ISYM=1	$f(\theta) = f(180+\theta) = -f(180-\theta)$	no constant terms  no cosine terms  no odd sine terms
ISYM=2 or 3	$f(\theta) = f(180-\theta) = -f(180+\theta)$	no constant term  no cosine terms  no even sine terms

Note, for example, in the case where the function is odd and ISYM=1, that though only the even sin terms are being calculated, some of these might also be zero. These numbers most likely will not be zero in the output but only small

numbers. The user must decide whether or not they are zero.

### Output

The entire input data is listed giving the FORTRAN name. The computed Fourier coefficients are written with a least-squares error for the curve fit up to that particular harmonic. A deck of cards containing the Fourier series coefficients is punched. The cards are punched in a form compatible with the input requirements of Program S3359.

If a plot is specified, a plot of the Fourier series curve fit is given on 12-inch grid paper. The actual curve is marked by X's, and the Fourier series curve fit is drawn with a solid line.

### Sample Input

PRESSURE LOAD 1;

1,1,.001,12,3,2,12;

0.,2.;

45.,2.;

90.,2.;

90,-2;

135.,-2;

180.,-2;

## APPENDIX C

### 370 OPERATING INSTRUCTIONS

**PROGRAM: STRESS ANALYSIS OF AN AXISYMMETRIC BODY  
WITH ASYMMETRIC LOADS**

## TAPES

Density	2	5	8	16	2	5	8	16	2	5	8	16	2	5	8	16	2	5	8	16
Data Set Name	SERVLIFE																			
Input/ Output	I	Ø			I	Ø			I	Ø			I	Ø			I	Ø		
Reserve List Scratch	R	L	SC		R	L	SC		R	L	SC		R	L	SC		R	L	SC	

NO

NO

**ERROR PROCEDURE:** Flush

**ESTIMATED MAXIMUM RUNNING TIME:** See user request

**SPECIAL INSTRUCTIONS:** None